

Plenary Session 4:

“Development, validation and application of a CFD based wind turbine simulation model”

Gabriel Usera - UdelaR

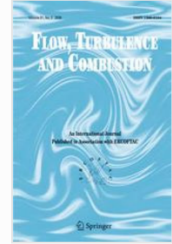


Increasing awareness of the potential of small/medium wind turbines

June 25-27, 2018
Huatulco, México



caffa3d MRRI - Open Source Ongoing Development



[Flow, Turbulence and Combustion](#)
October 2008, 81:471 | [Cite as](#)

A Parallel Block-Structured Finite Volume Method for Flows in Complex Geometry with Sliding Interfaces

Authors [Authors and affiliations](#)

G. Usera , A. Vernet, J. A. Ferré

2004 ... 2008



[Cluster Computing](#)

June 2014, Volume 17, [Issue 2](#), pp 231–241 | [Cite as](#)

A general purpose parallel block structured open source incompressible flow solver

Authors [Authors and affiliations](#)

Mariana Mendina, Martin Draper, Ana Paula Kelm Soares, Gabriel Narancio, Gabriel Usera 

2014

[Ann Biomed Eng.](#) 2015 Jan;43(1):154-67. doi: 10.1007/s10439-014-1082-9. Epub 2014 Aug 13.

Accuracy and reproducibility of patient-specific hemodynamic models of stented intracranial aneurysms: report on the Virtual Intracranial Stenting Challenge 2011.

2015

[Cito S¹](#), [Geers AJ](#), [Arroyo MP](#), [Palero VR](#), [Pallarés J](#), [Vernet A](#), [Blasco J](#), [San Román L](#), [Fu W](#), [Qiao A](#), [Janiga G](#), [Miura Y](#), [Ohta M](#), [Mendina M](#), [Usera G](#), [Frangi AF](#).

[Computers and Fluids 156 \(2017\) 200–208](#)



Contents lists available at [ScienceDirect](#)

Computers and Fluids

journal homepage: www.elsevier.com/locate/compfluid



2017



Coupled discrete element and finite volume methods for simulating loaded elastic fishnets in interaction with fluid

Paolo Sassi, Jorge Freiria, Paula La Paz, Mariana Mendina, Martin Draper, Gabriel Usera*

IMFIA - UdelaR - J. Herrera y Reissig 565, Montevideo, 11300, Uruguay

Tenth International Conference on Computational Fluid Dynamics (ICCFD10), Barcelona, Spain, July 9-13, 2018

ICCFD10-2018-xxxx

2018 ...

A Large Eddy Simulation model for the study of wind turbine interactions and its application

M. Draper^{1,*}, A. Guggeri¹, M. Mendina¹, G. Usera¹, and F. Campagnolo²

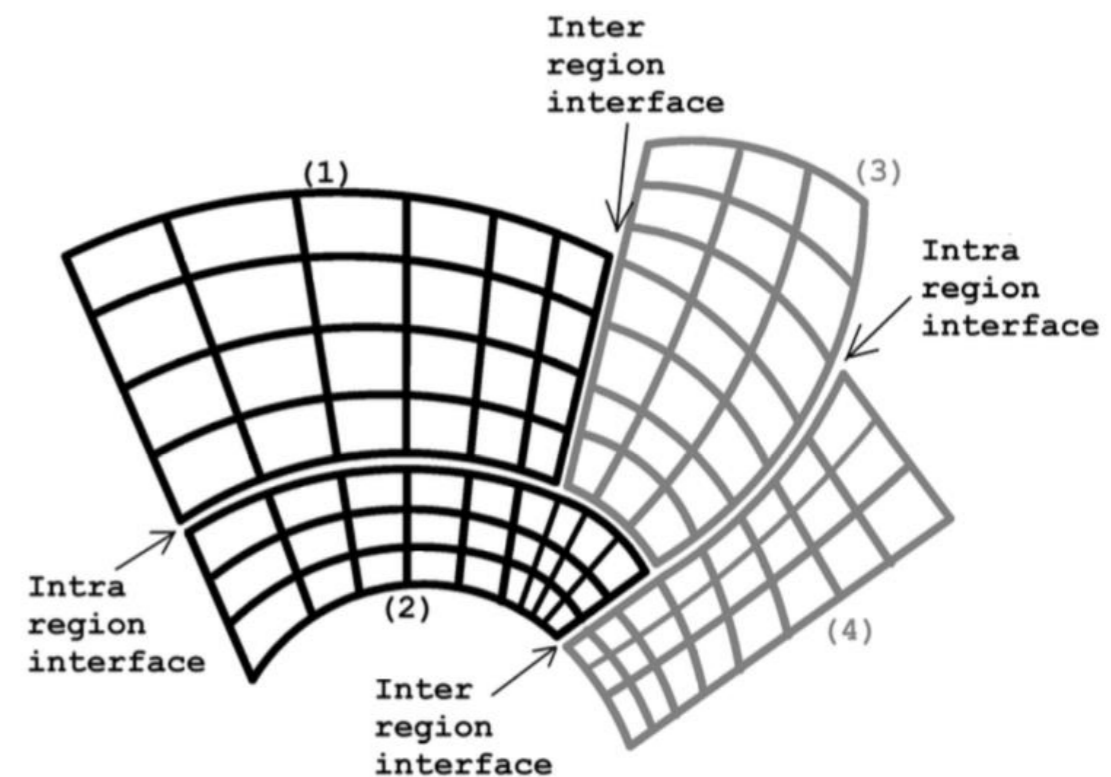
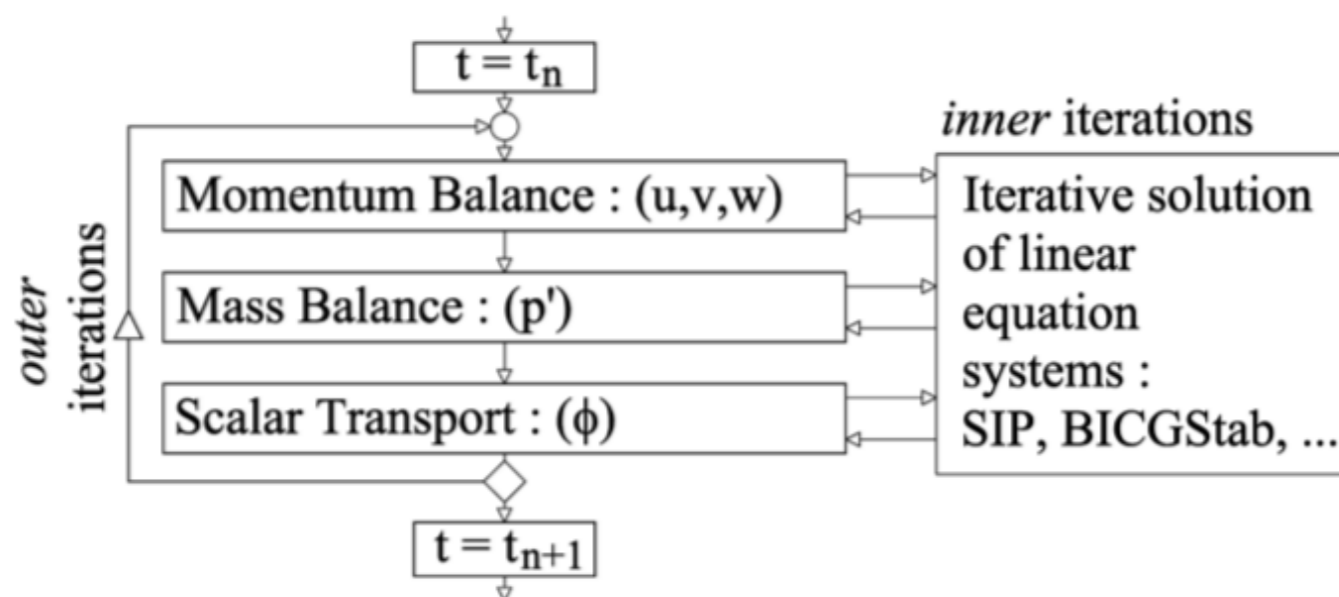
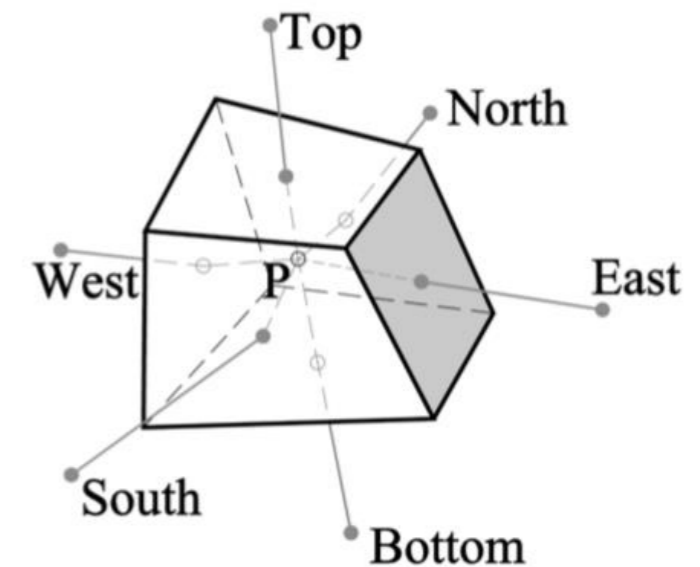
caffa3d.MBRi : Finite Volume Flow Solver

$$\int_S (\mathbf{v} \cdot \hat{\mathbf{n}}_S) dS = 0$$

$$\int_{\Omega} \rho \frac{\partial u}{\partial t} d\Omega + \int_S \rho u (\mathbf{v} \cdot \hat{\mathbf{n}}_S) dS =$$

$$\int_{\Omega} \rho \beta (T - T_{ref}) \mathbf{g} \cdot \hat{\mathbf{e}}_1 d\Omega + \int_S -p \hat{\mathbf{n}}_S \cdot \hat{\mathbf{e}}_1 dS +$$

$$\int_S (2\mu \mathbf{D} \cdot \hat{\mathbf{n}}_S) \cdot \hat{\mathbf{e}}_1 dS$$



(Mendina et al, 2014)

Immersed Boundary Conditions



$$\int_{\Omega} \rho \frac{\partial \vec{v}}{\partial t} d\Omega + \int_S \rho \vec{v} (\vec{v} \cdot \hat{n}_S) dS = \int_{\Omega} \rho \beta (T - T_{ref}) \vec{g} d\Omega + \int_S -p \hat{n}_S dS + \int_S (2\mu \mathbf{D} \cdot \hat{n}_S) dS + \mathbf{F}_{imb}$$

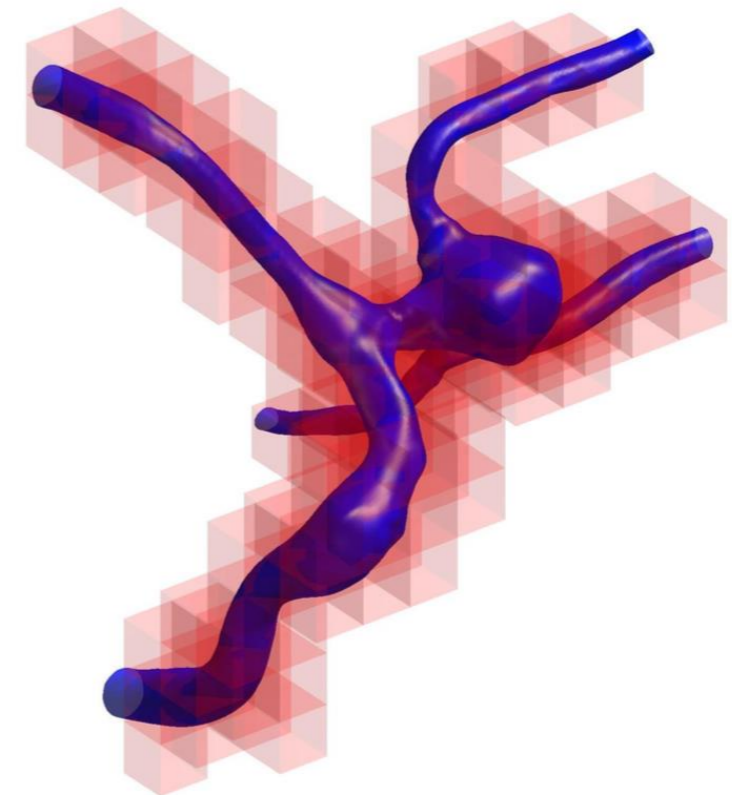
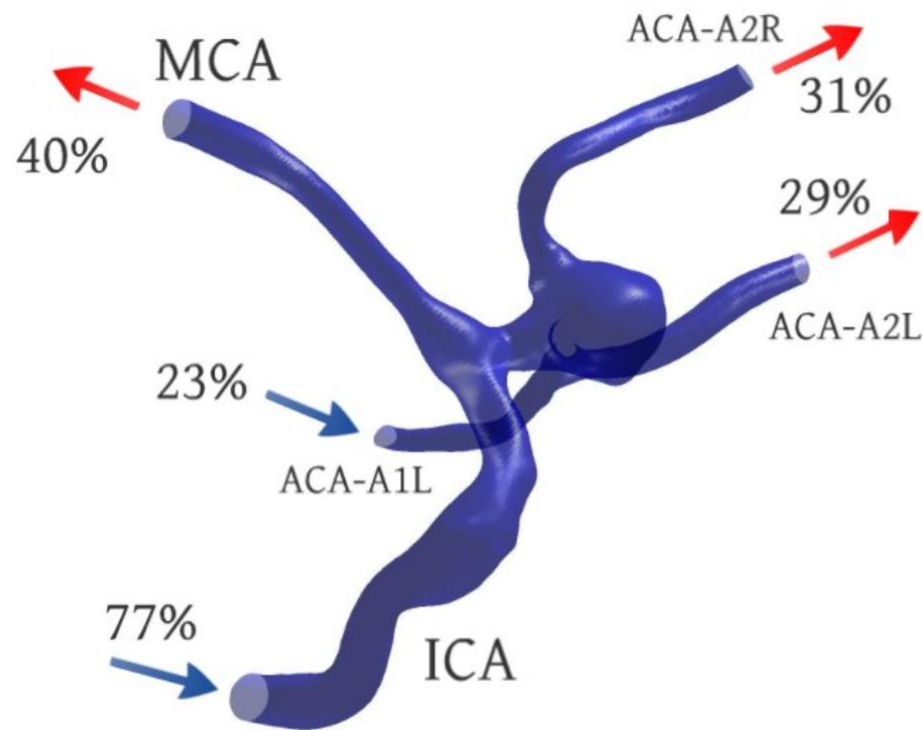


Fig. 14 Patient specific aneurysm geometry with specified flow distribution at inlets and outlets.

(Mendina et al, 2014)

Immersed Boundary Conditions

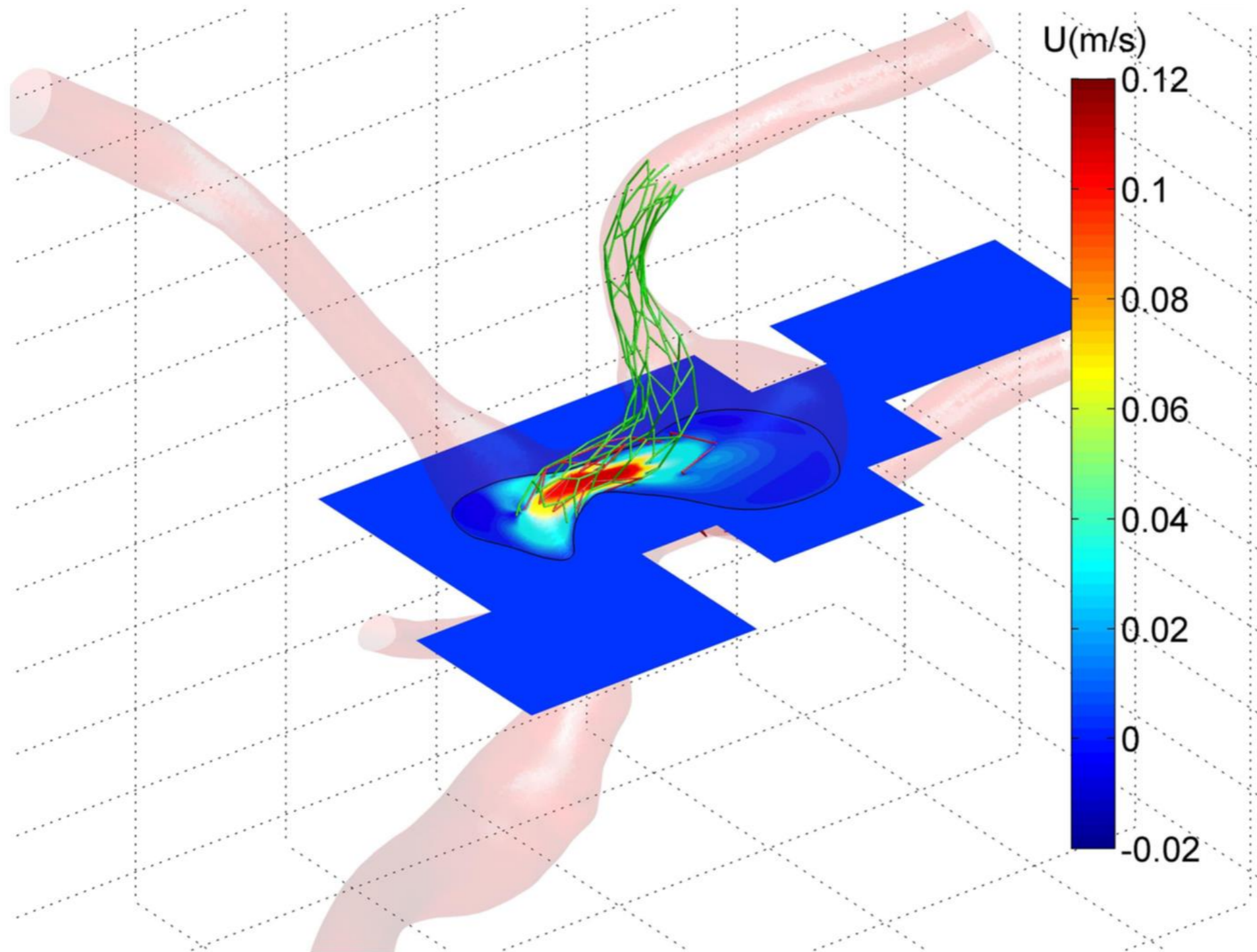


Fig. 16 Detail of flow interaction with stents.

Multi body dynamics



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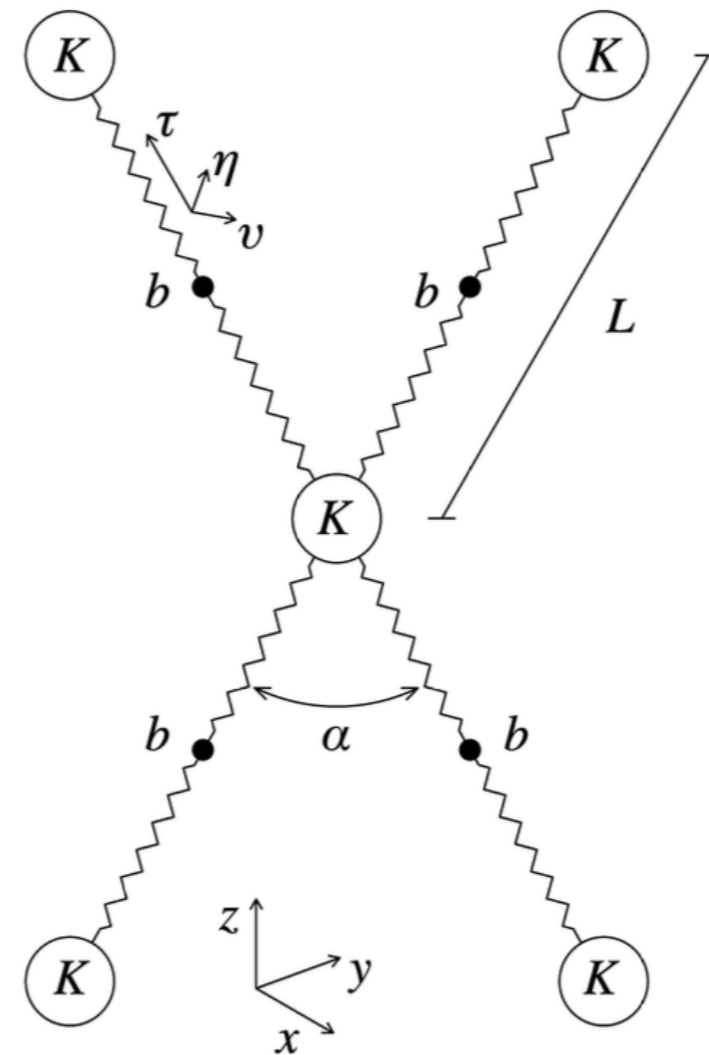
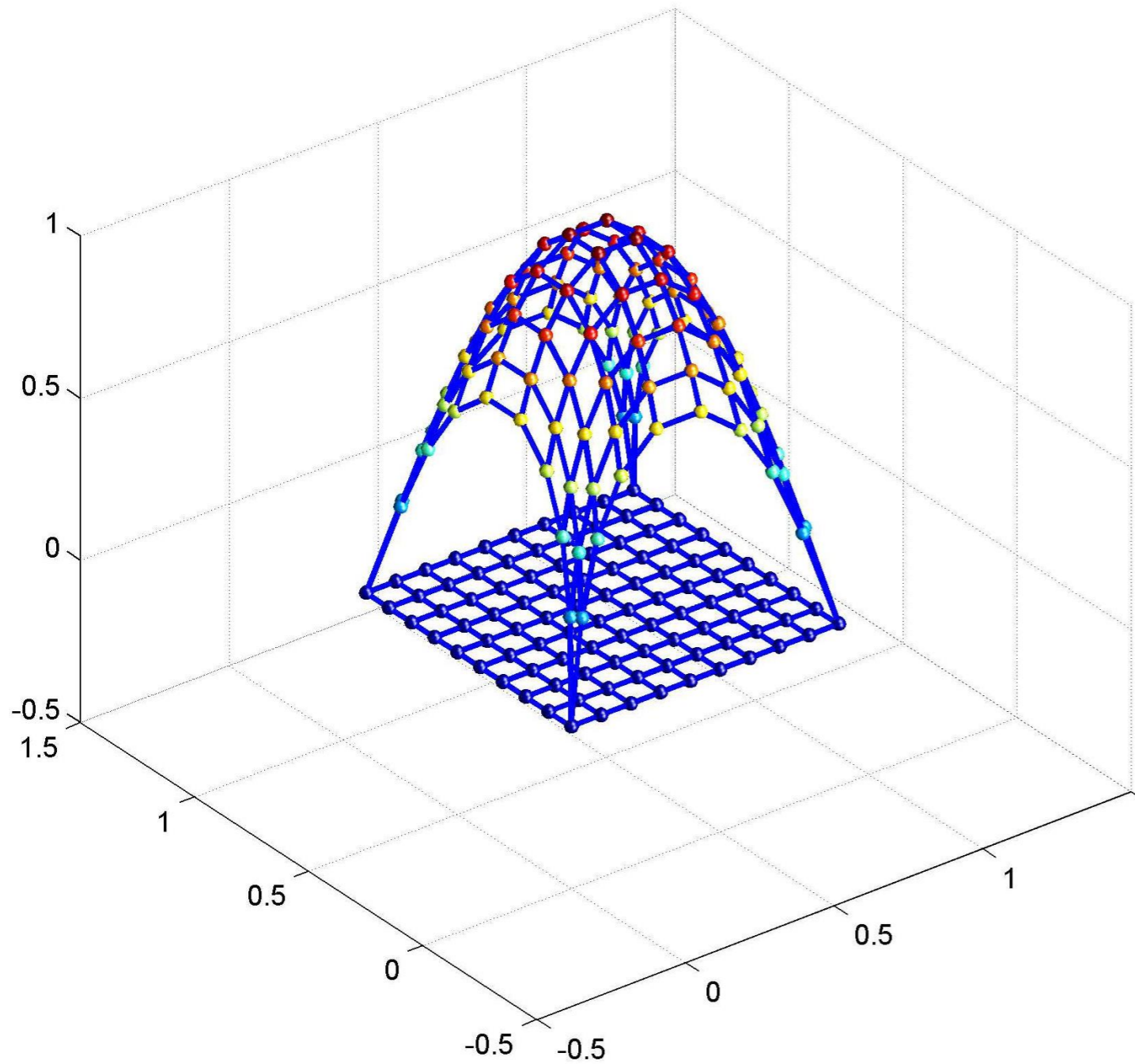


Figure 1: Sketch of the point masses and massless strings.

(Sassi et al, 2017)

Multi body dynamics

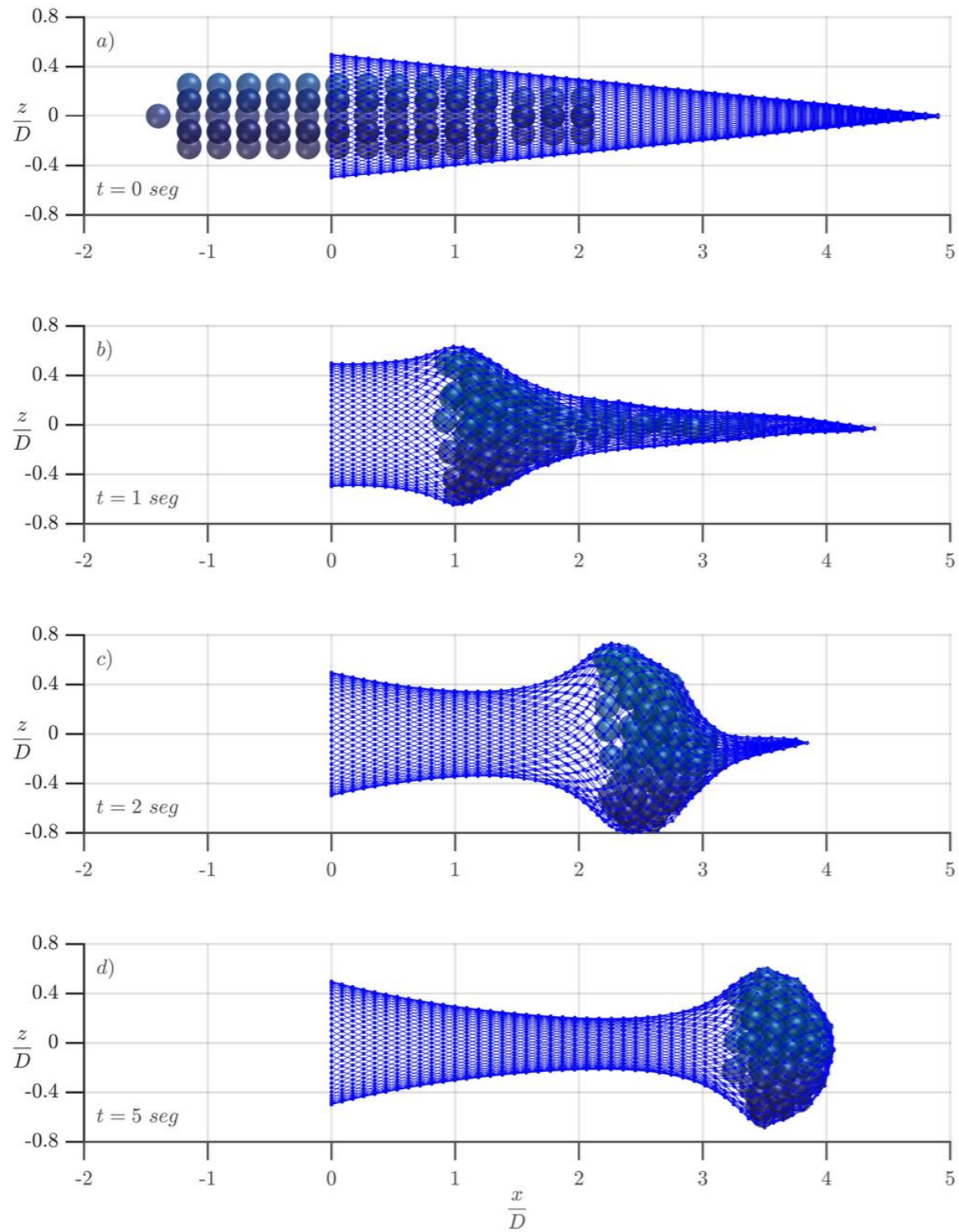


Figure 3: Evolution of the fishnet shape, at time steps $t = 0, 1.0, 2.0, 5.0$ s, as the load is dragged into the fishnet.

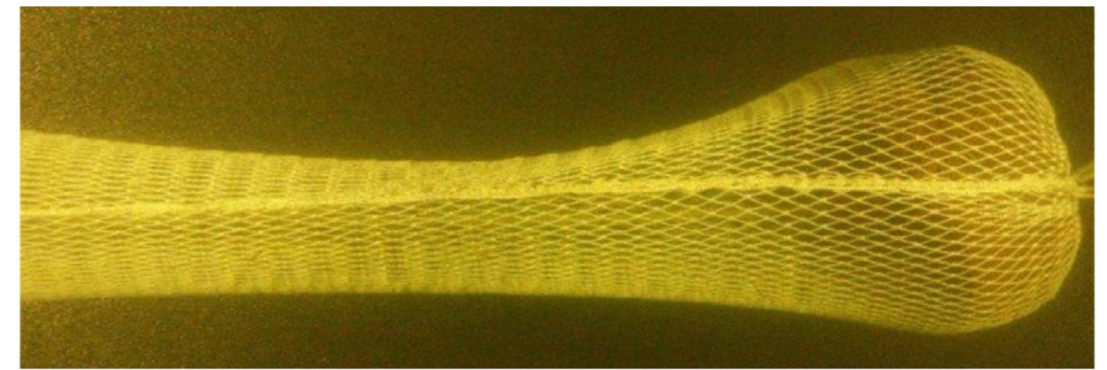


Fig. 5. Shape of the fishnet during the towing tank test.

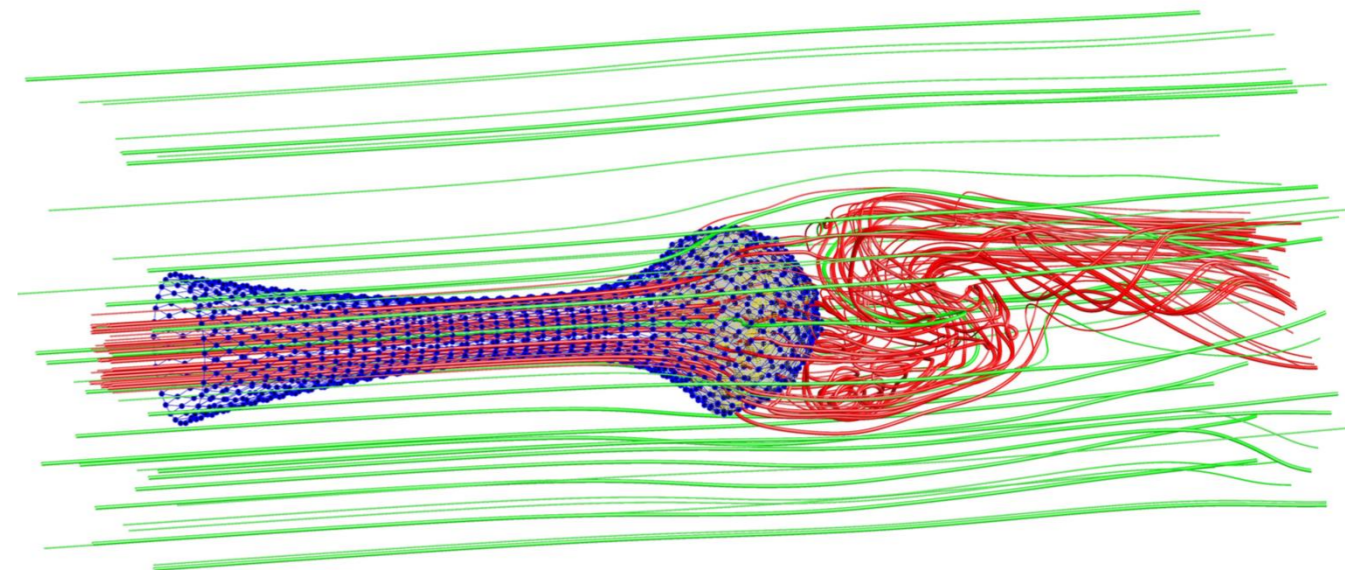
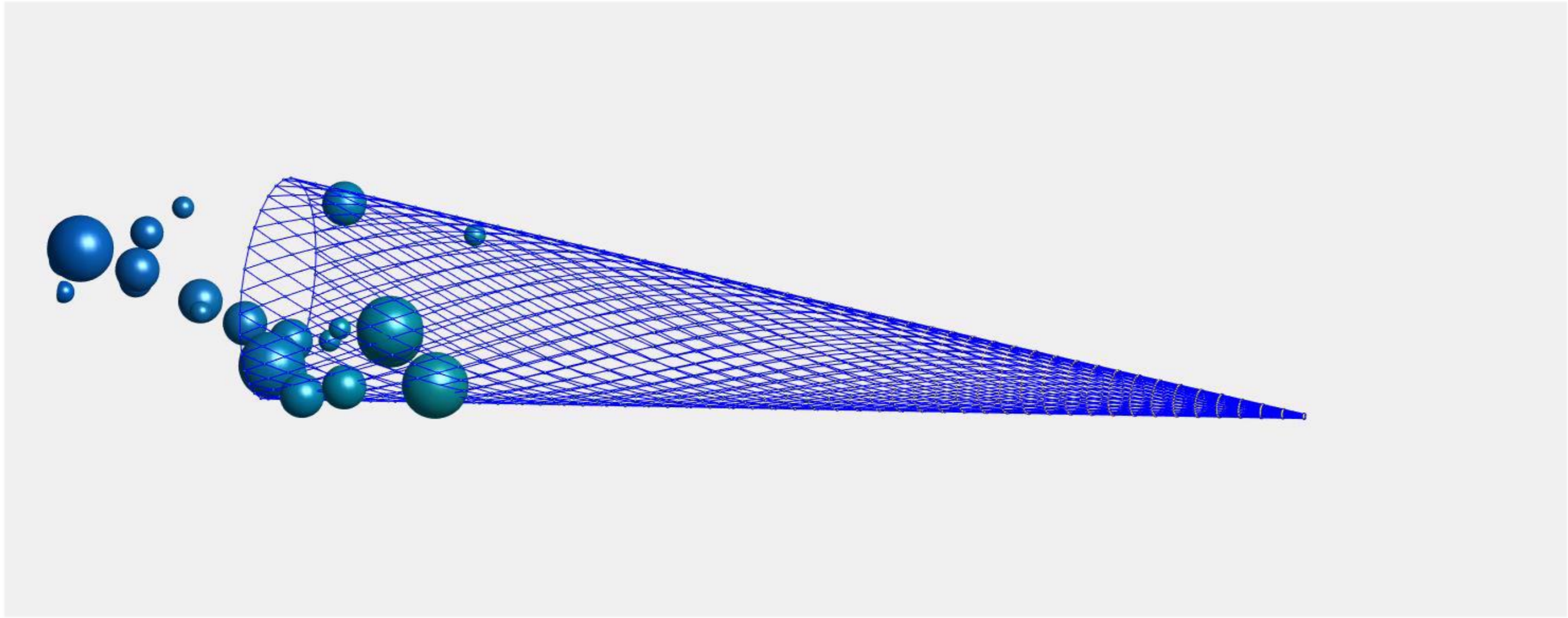


Figure 5: Stream Lines

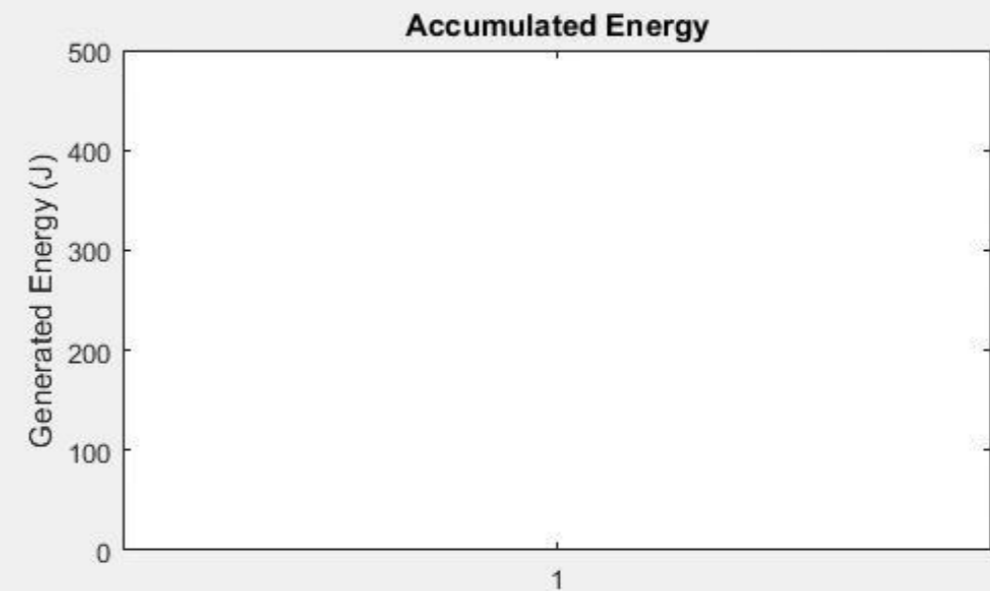
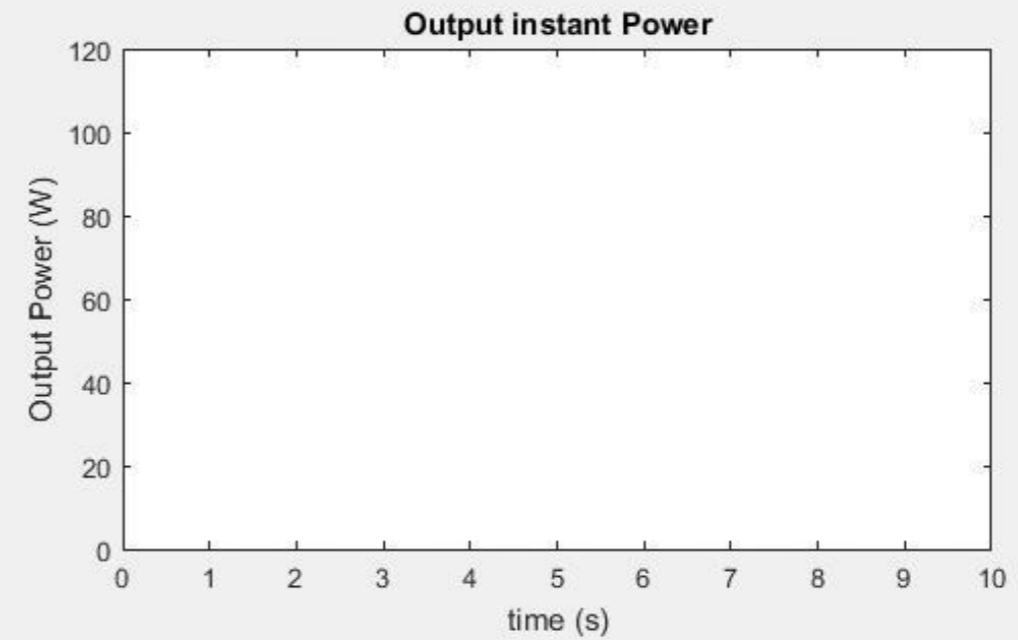
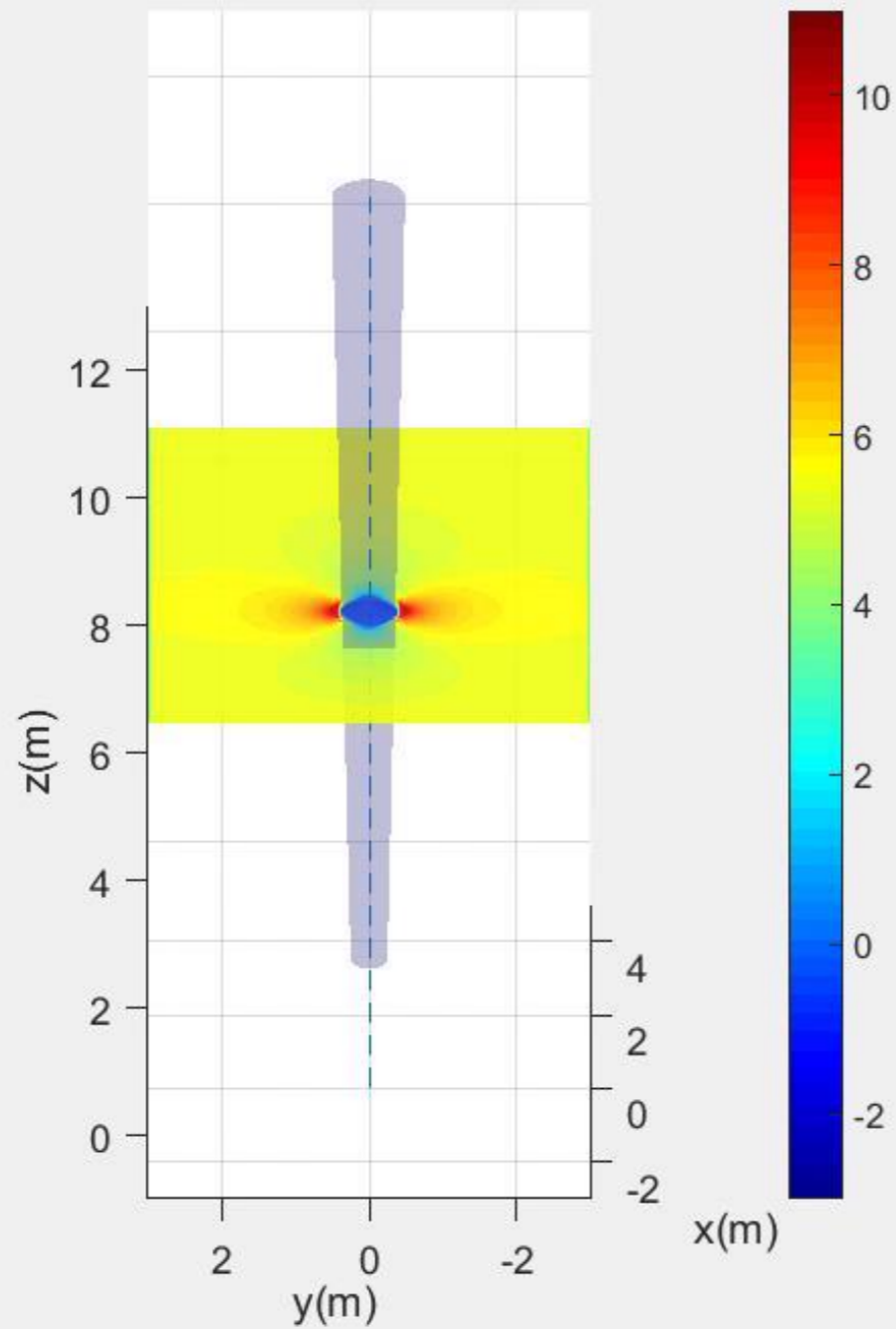
(Sassi et al, 2017)

Loaded Fishnet Simulation



Vorticity Wind Turbine

Vortex oscilando, $U_0=5.5\text{m/s}$, $t=0.04$



CFD BEM WTG model

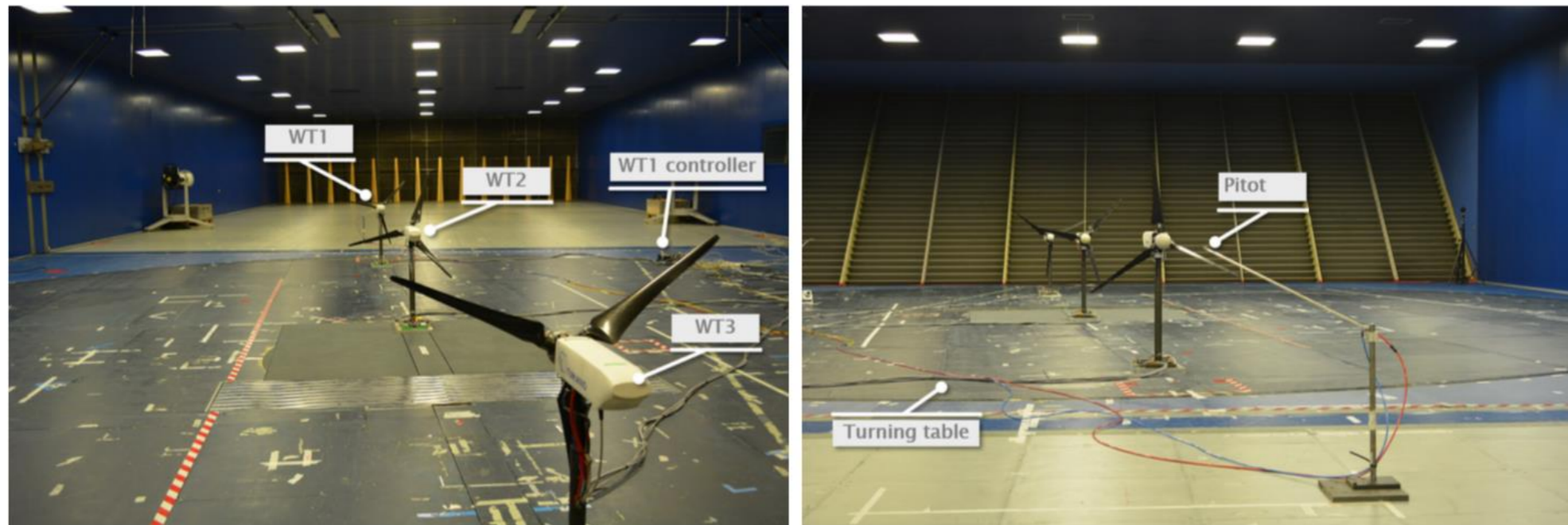


Figure 3: Wind farm layout in the wind tunnel.

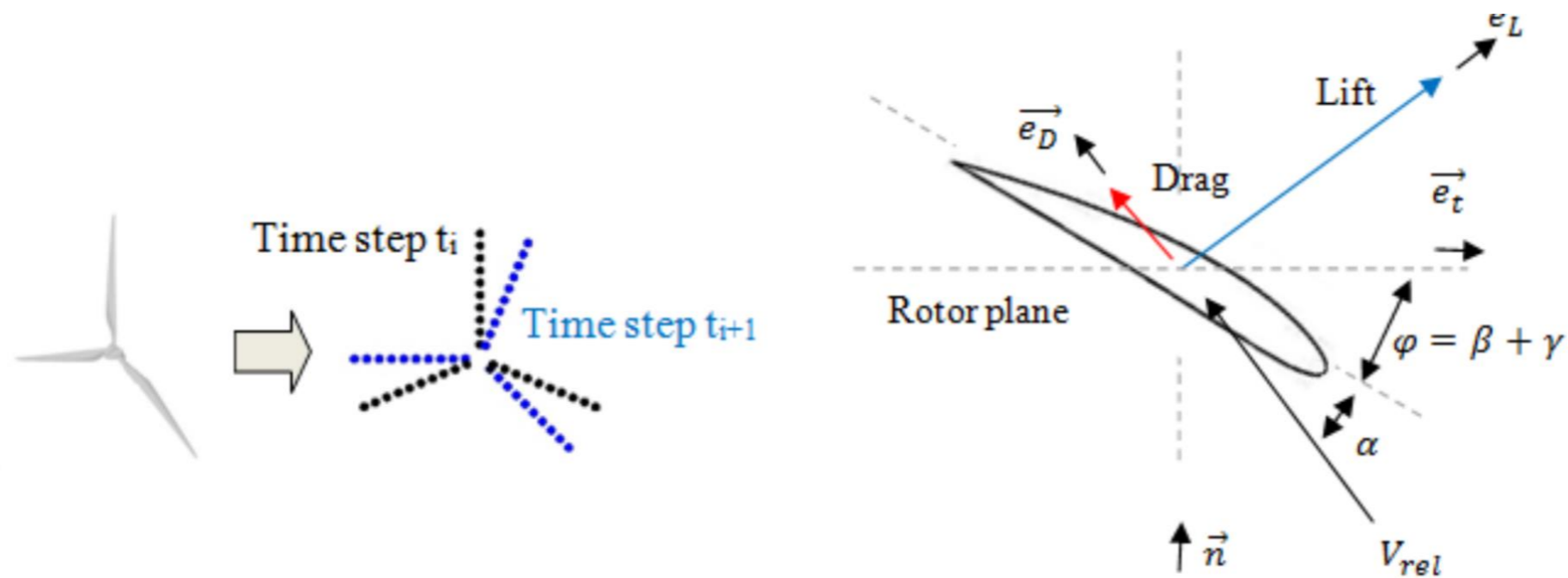


Figure 2: ALM rotor representation (left) and a cross-sectional airfoil radial section (right).

(Draper et al, 2018)

CFD BEM WTG model

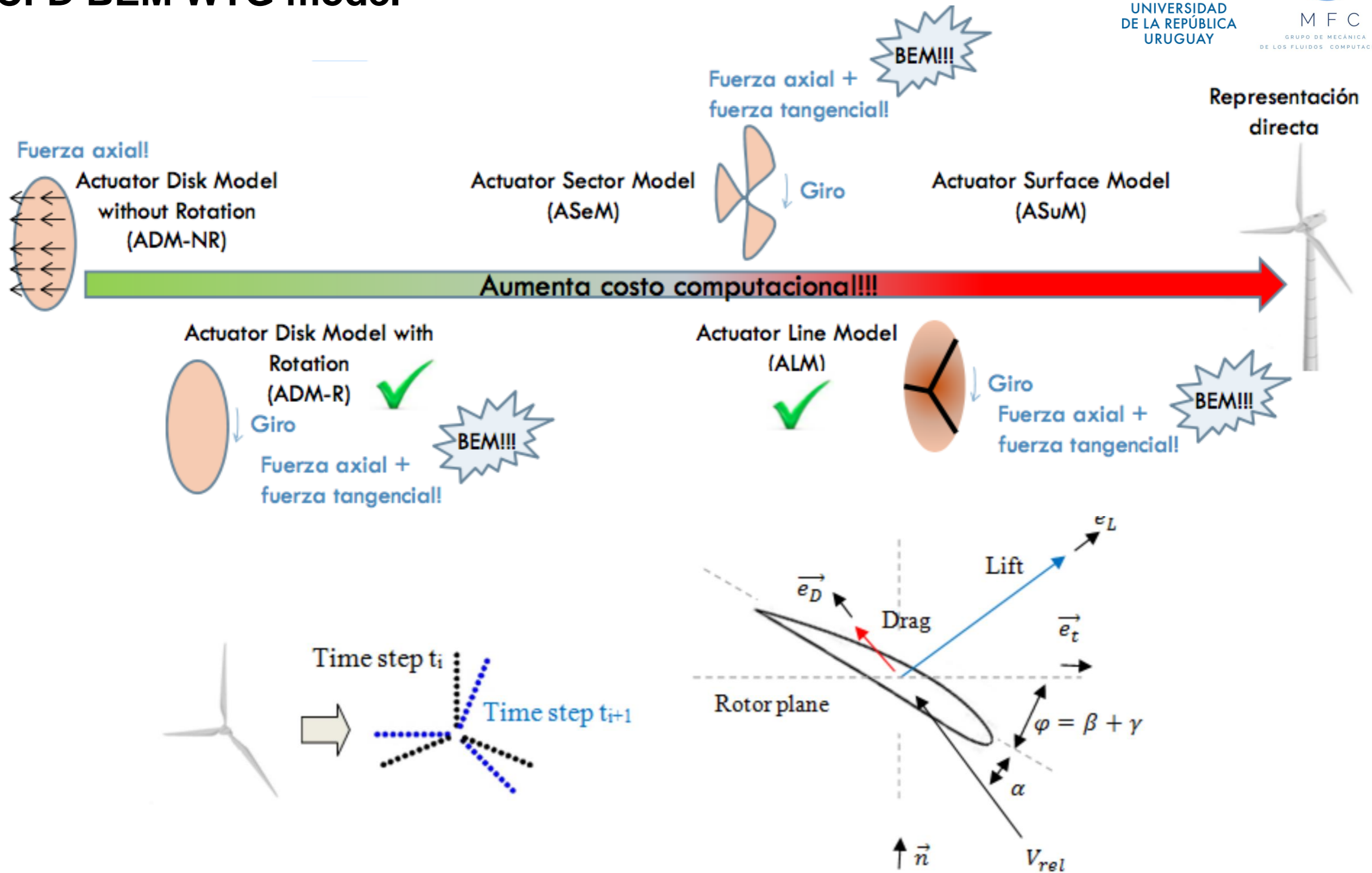


Figure 2: ALM rotor representation (left) and a cross-sectional airfoil radial section (right).

(Draper et al, 2018)

CFD BEM WTG model

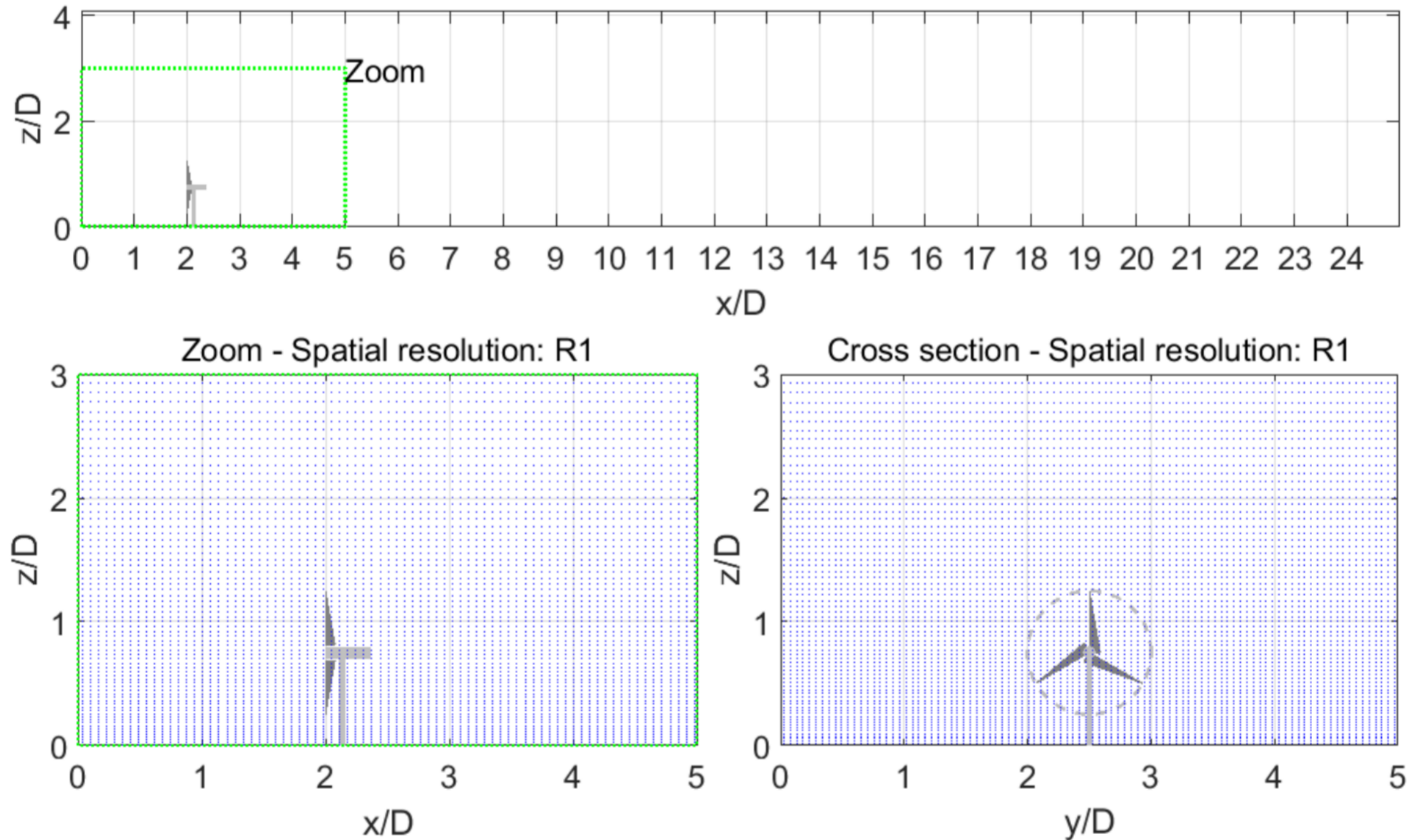


Figure 5: Side view of the computational domain (top). Spatial resolution R1: zoom view close to model wind turbine (bottom, left) and cross section (bottom, right). Blue dots represent grid node centers.

CFD BEM WTG model

Table 1: Numerical setup.

Spatial resolution	N_x	N_y	N_z	$\Delta x(m)$	$\Delta y(m)$	$z_{min}(m)$	$R/\Delta x$	$R/\Delta y$	Nz_{Rotor}
R0	256	64	64	0.107	0.086	0.035	5.1	6.4	22
R1	384	96	80	0.072	0.057	0.022	7.7	9.6	30
R2	512	128	108	0.054	0.043	0.016	10.2	12.8	40

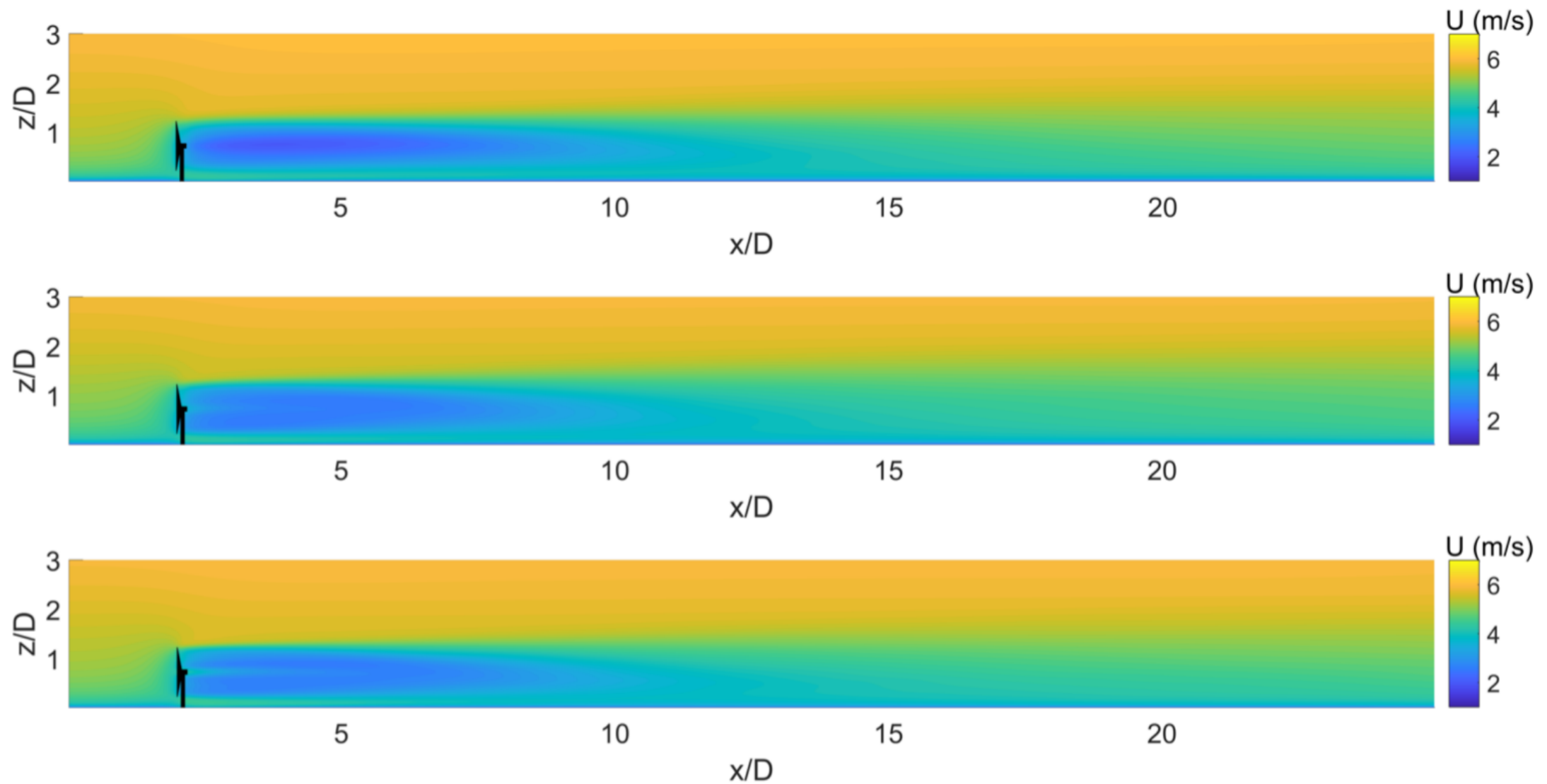


Figure 6: Mean streamwise velocity component in a plane passing through the rotor center. Top: spatial resolution R0, center: spatial resolution R1, bottom: spatial resolution R2. The model wind turbine is sketched in black.

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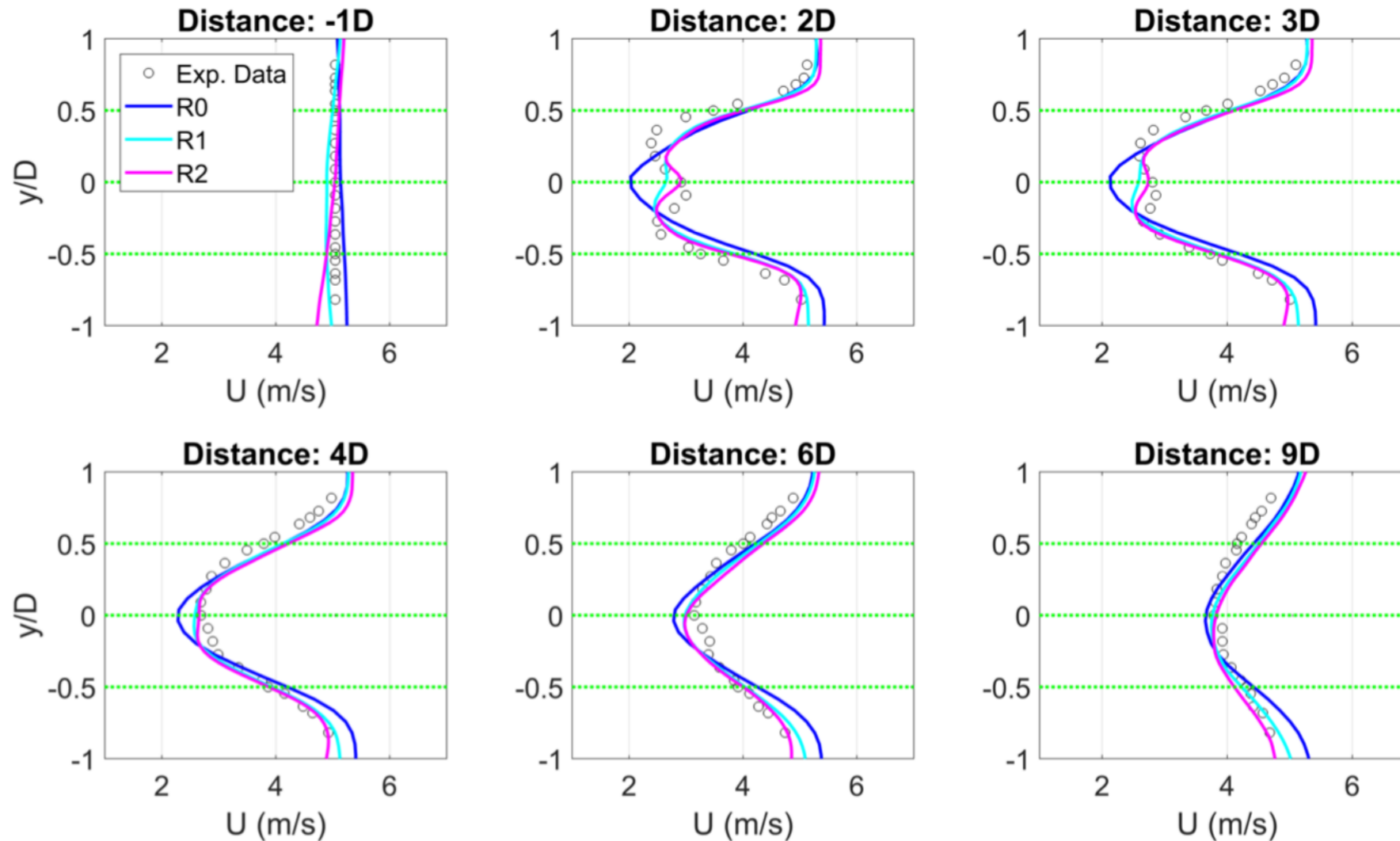


Figure 7: Mean streamwise velocity component at different locations in the wake at hub height, for the three spatial resolutions considered. Dotted green lines represent the rotor center and blade tips. Open circles represent experimental data.

Table 2: Power and thrust coefficients. Values in brackets represent the difference between experimental and simulated value.

		Cp	Ct
Exp. Data		0.453	0.788
Sim.	R0	0.451 (-0.3%)	0.717 (-9.0%)
	R1	0.428 (-5.4%)	0.716 (-9.2%)
	R2	0.414 (-8.5%)	0.699 (-11.3%)

(Draper et al, 2018)

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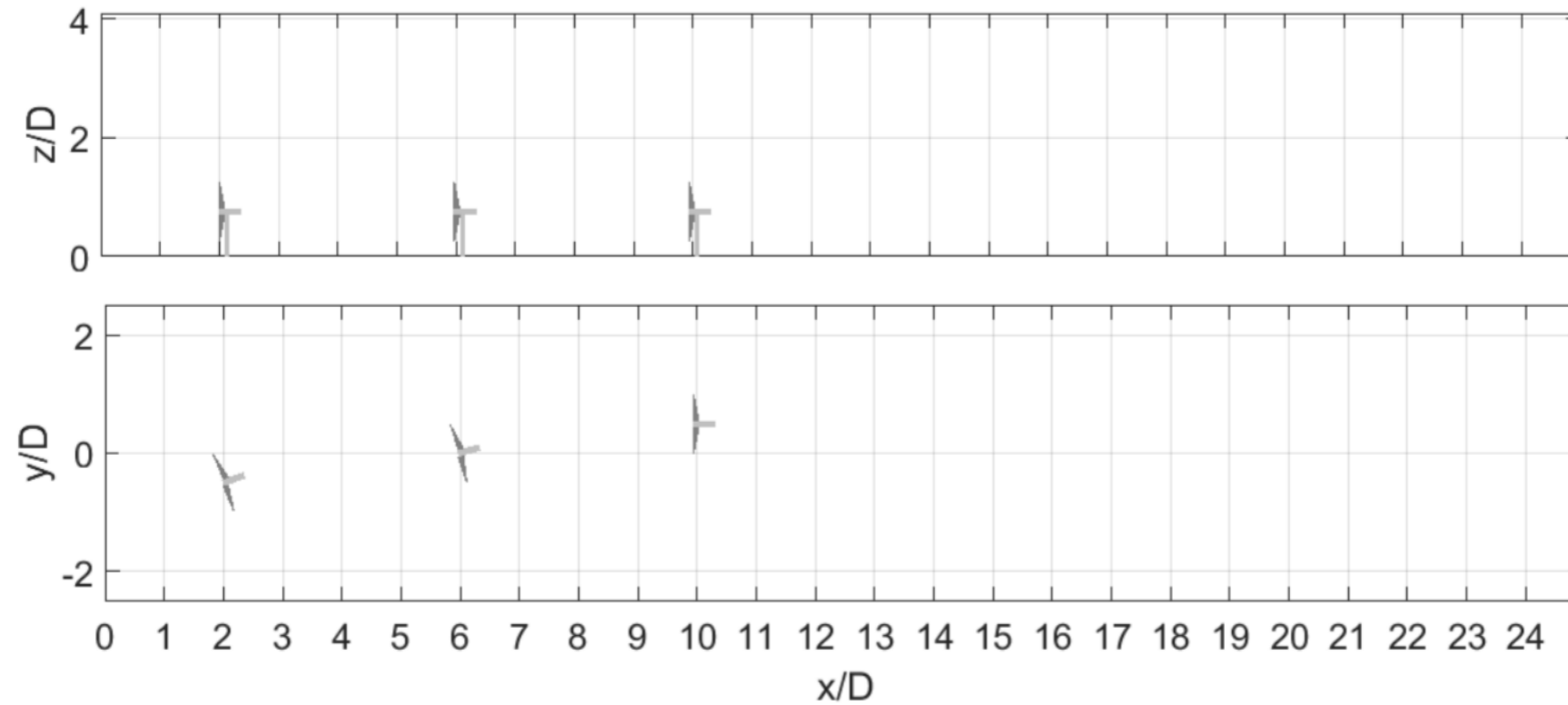


Figure 10: Computational domain showing the layout for the selected yaw setting. The wind turbine models are sketched in grey.

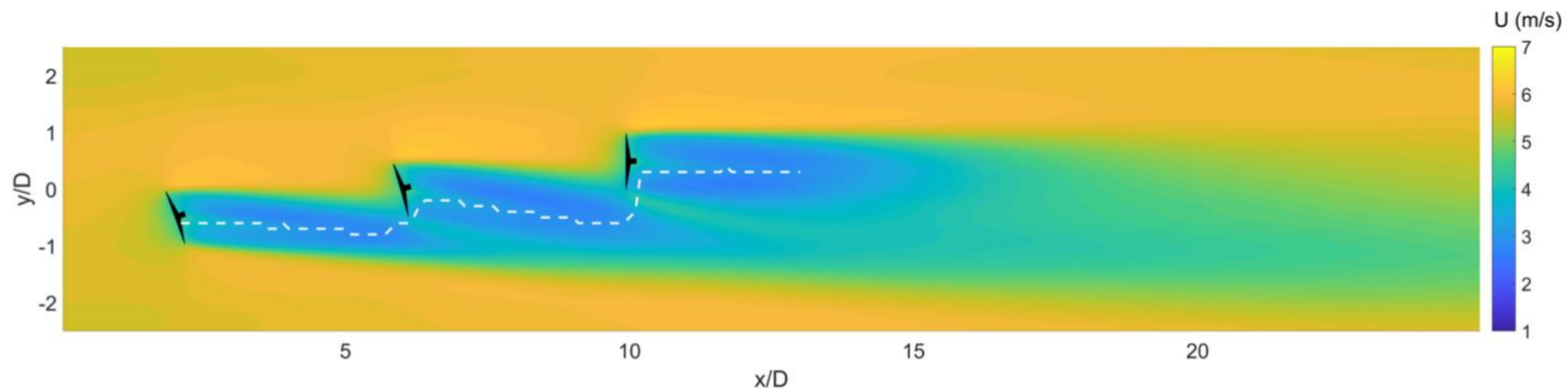


Figure 11: Mean streamwise velocity component in a horizontal plane passing 0.10 m above the rotor center. Spatial resolution R1. The model wind turbines are sketched in black. The white dash line represents the wake center computed from the experimental data by minimization of Equation 7.

CFD BEM WTG model

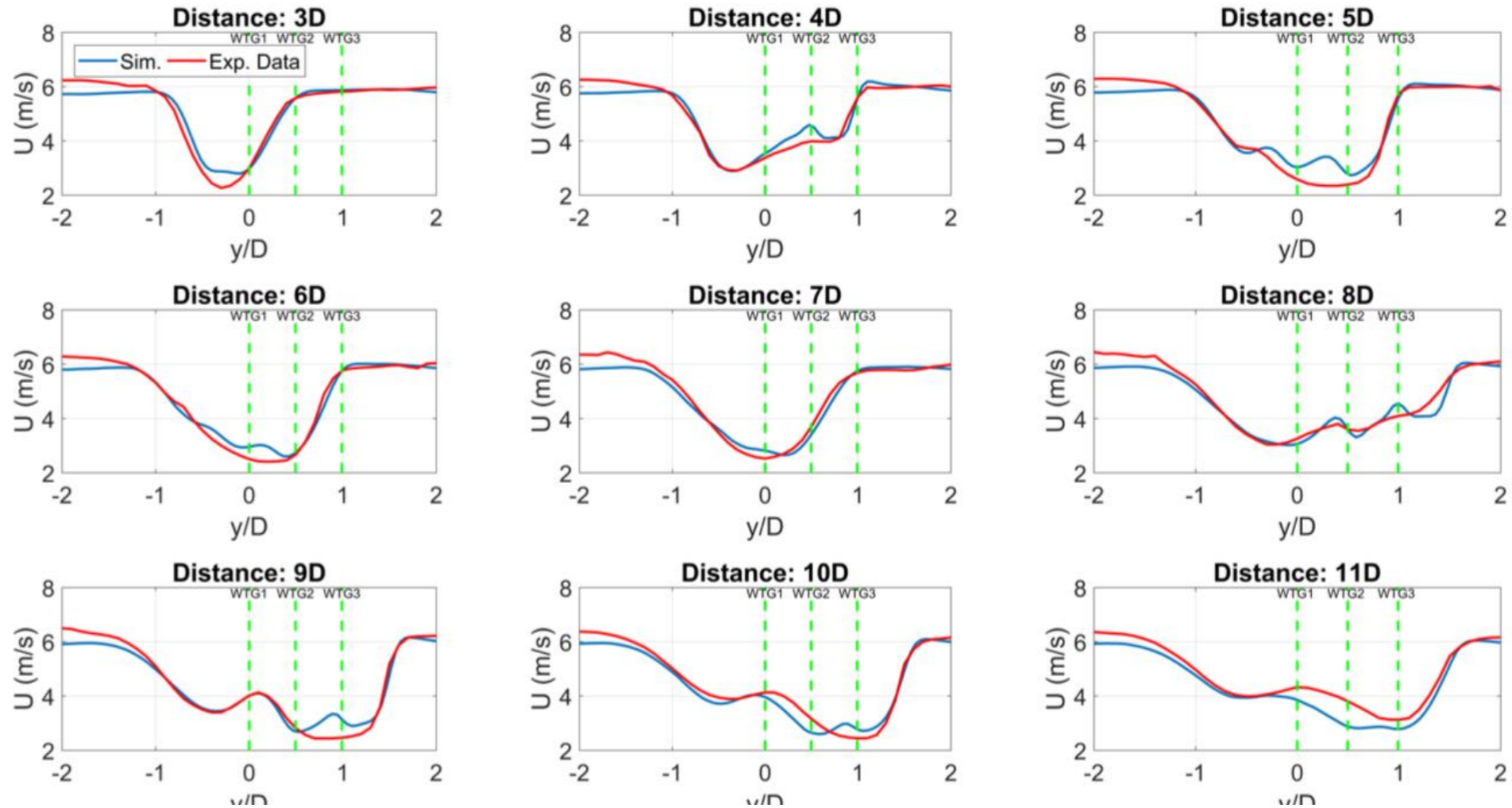


Figure 12: Mean streamwise velocity component in a horizontal plane 0.10 m above the rotor center at different locations in the wake. Spatial resolution R1. The wind turbine rotor centers are represented with green dashed lines. Distance is measured from the rotor plane of the upwind model wind turbine.

Table 3: Power coefficient of each model wind turbine.

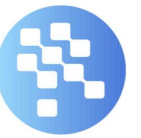
	WTG1	WTG2	WTG3	Total
Exp. Data	0.388	0.350	0.404	1.142
Sim.	0.400	0.390	0.462	1.251
Cp diff.	3.1%	11.3%	14.4%	9.6%

(Draper et al, 2018)

CFD BEM WTG model



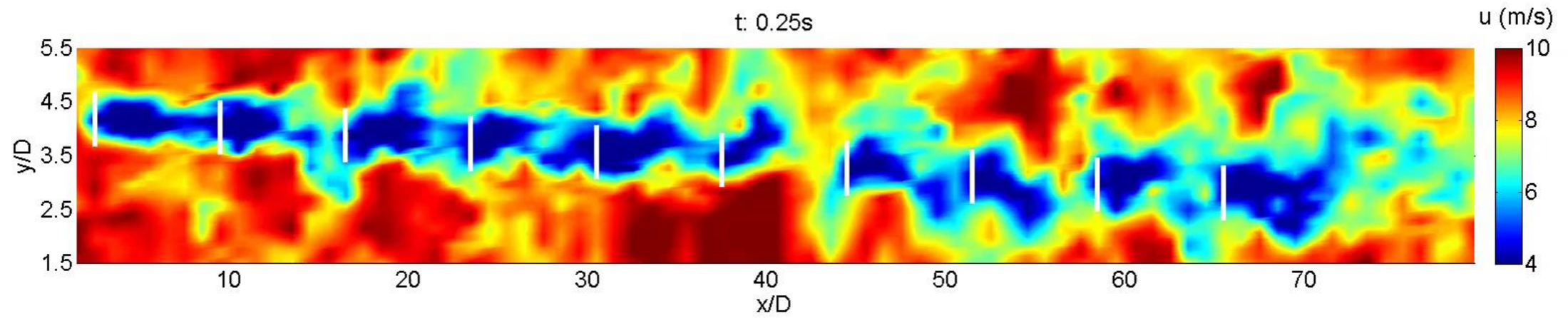
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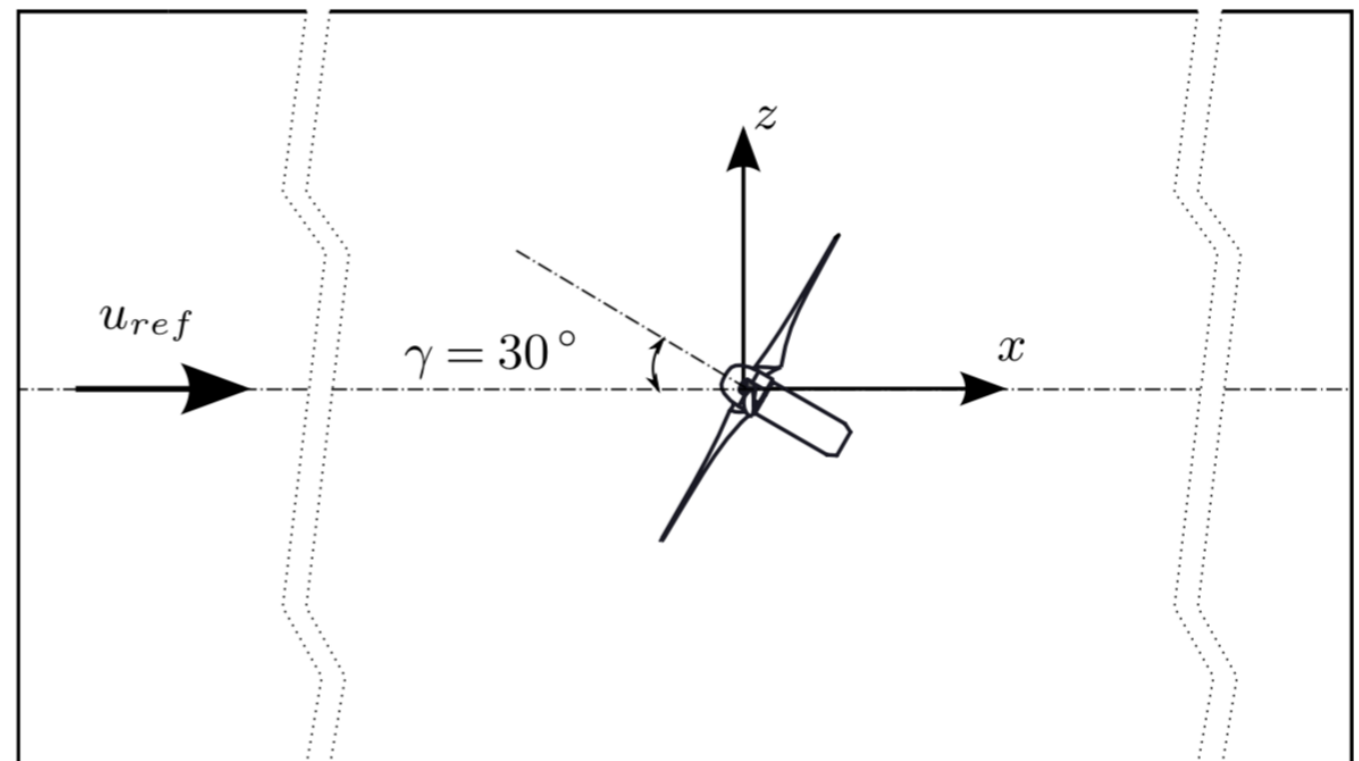
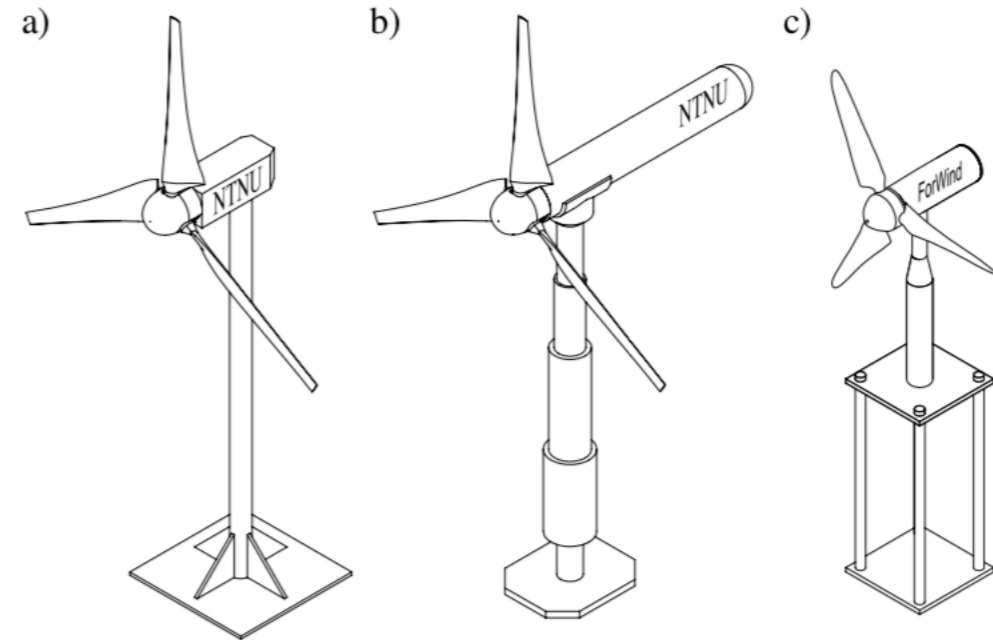
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Blind Test #5 - 2017



Blind Test #5 - 2017

Table 2. Overview of simulation methods and parameters. Abbreviations: Improved Delayed Detached Eddy Simulation (IDDES), Large Eddy Simulation (LES), Actuator Line (ACL), Fully Resolved (FR).

Participant	Simulation code	Flow solver type	Rotor model	Airfoil polars	Tower, nacelle	Mesh properties	Number of cells
Siemens	Star-CCM+	IDDES	FR	-	FR	Hexah./polyh.	$\approx 30.0 \cdot 10^6$
POLIMI	ALEVMM	LES	ACL	X-Foil	No	Cartesian	$\approx 4.1 \cdot 10^6$
UdelaR	caffa3d	LES	ACL	X-Foil	Yes	Cartesian	$\approx 0.7 \cdot 10^6$
KTH	Nek5000	LES	ACL	Experiments	Yes	Uniform	$\approx 0.08 \cdot 10^6$

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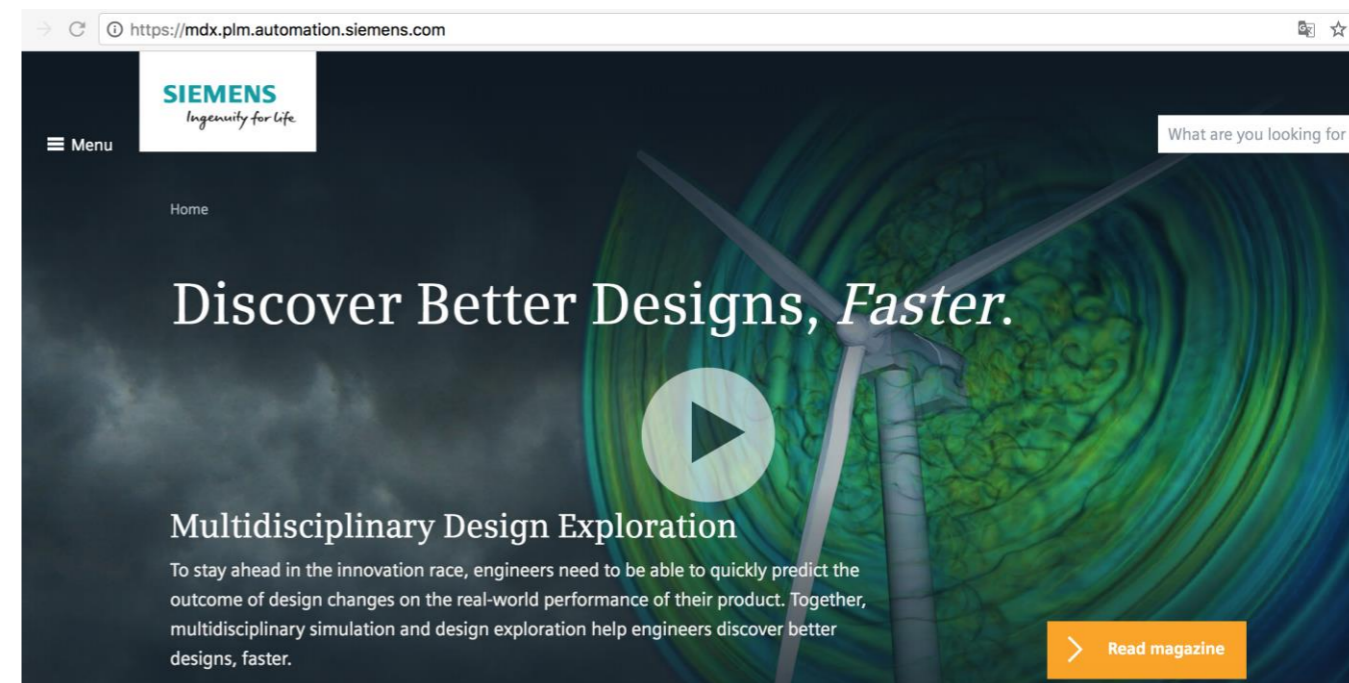
Siemens to buy CD-adapco for close to \$1 billion: source

Liana B. Baker

3 MIN READ



(Reuters) - Siemens AG (SIEGn.DE), Europe's biggest industrial group, has agreed to buy CD-adapco, a privately held U.S. engineering software firm, for close to \$1 billion in cash, according to a person familiar with the matter.



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(Mühle et al, 2018)

Blind Test #5 - 2017

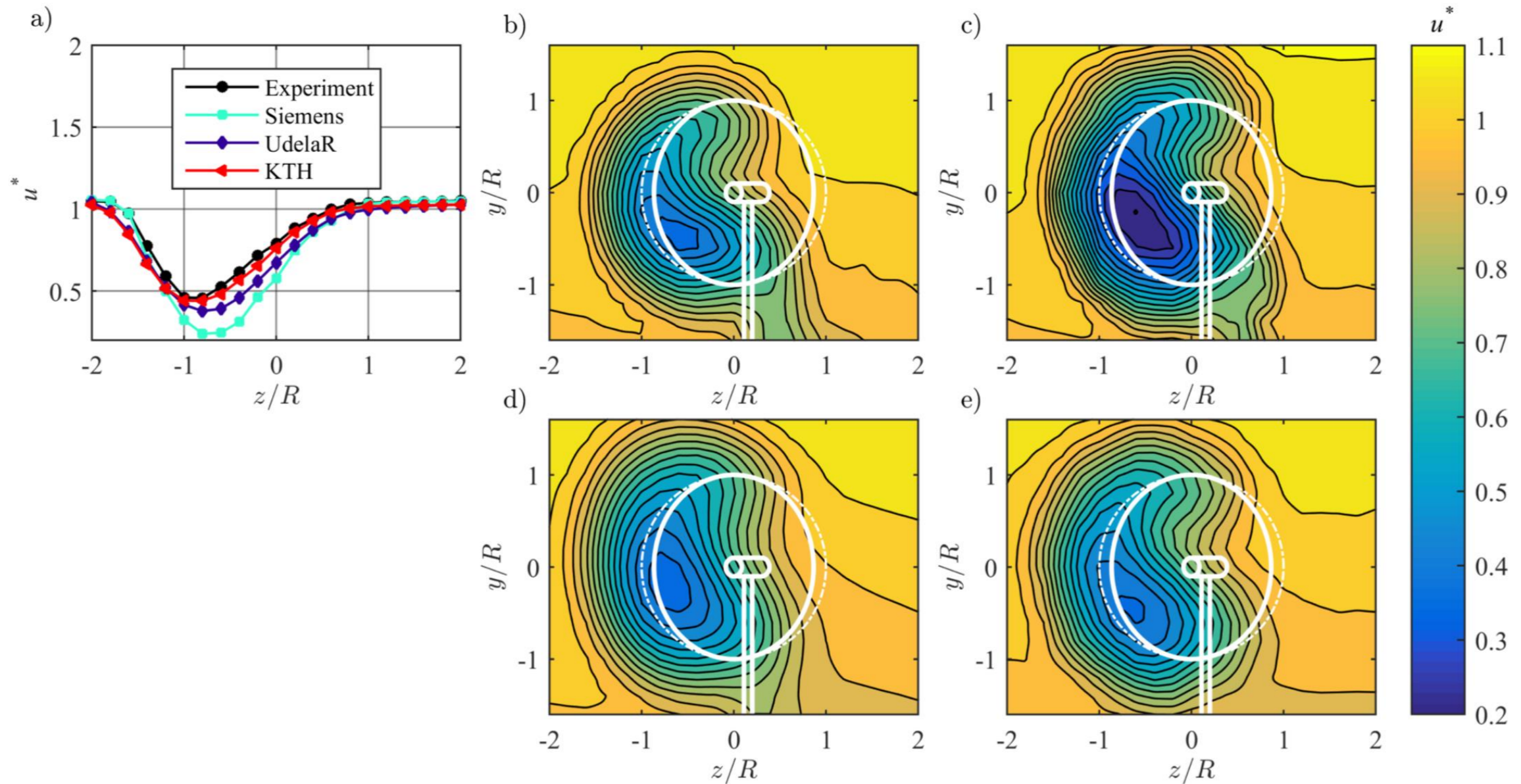


Figure 13. (a) Line plots and (b-e) contour plots for normalized streamwise mean velocity u^* in the wake 3D behind upstream ForWind turbine, from (b) experiments, (c) Siemens, (d) UdelaR and (e) KTH and. The white lines represent the turbine rotor, nacelle and tower, solid lines $\gamma = 30^\circ$, dashed lines $\gamma = 0^\circ$.

Blind Test #5 - 2017

Table 6. Comparison parameters: Skew angle (ξ), wake deflection (δ) and available power in the wake (P_{wake}^*) and their differences to the measurements. Statistical performance measures: $NMSE$ and r for u^* , v^* and k^* at $3D$ and $6D$ behind upstream ForWind turbine.

	Institution	Skew angle	Deflection (z/R)	Difference (z/R)	P_{wake}^* [-]	Difference [%]	$NMSE_u$	r_u	r_v	$NMSE_k$	r_k
$3D$	Experiments	4.10°	-0.429		0.285						
	Siemens	3.71°	-0.388	0.041	0.141	-49.4%	0.012	0.968	0.813	0.383	0.889
	UdelaR	4.88°	-0.510	-0.082	0.207	-27.6%	0.007	0.953	0.802	0.734	0.878
	KTH	5.27°	-0.551	-0.122	0.233	-18.%	0.005	0.960	0.851	0.202	0.905
$6D$	Experiments	3.80°	-0.796		0.533						
	Siemens	3.41°	-0.714	0.082	0.430	-19.3%	0.002	0.960	0.845	0.047	0.961
	UdelaR	4.00°	-0.837	-0.041	0.540	1.2%	0.001	0.963	0.799	0.067	0.956
	KTH	4.19°	-0.878	-0.082	0.475	-11.0%	0.002	0.950	0.884	0.052	0.947

SWTOMP "Fence Experiment", *back of the envelope idea*



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-----Original Appointment-----

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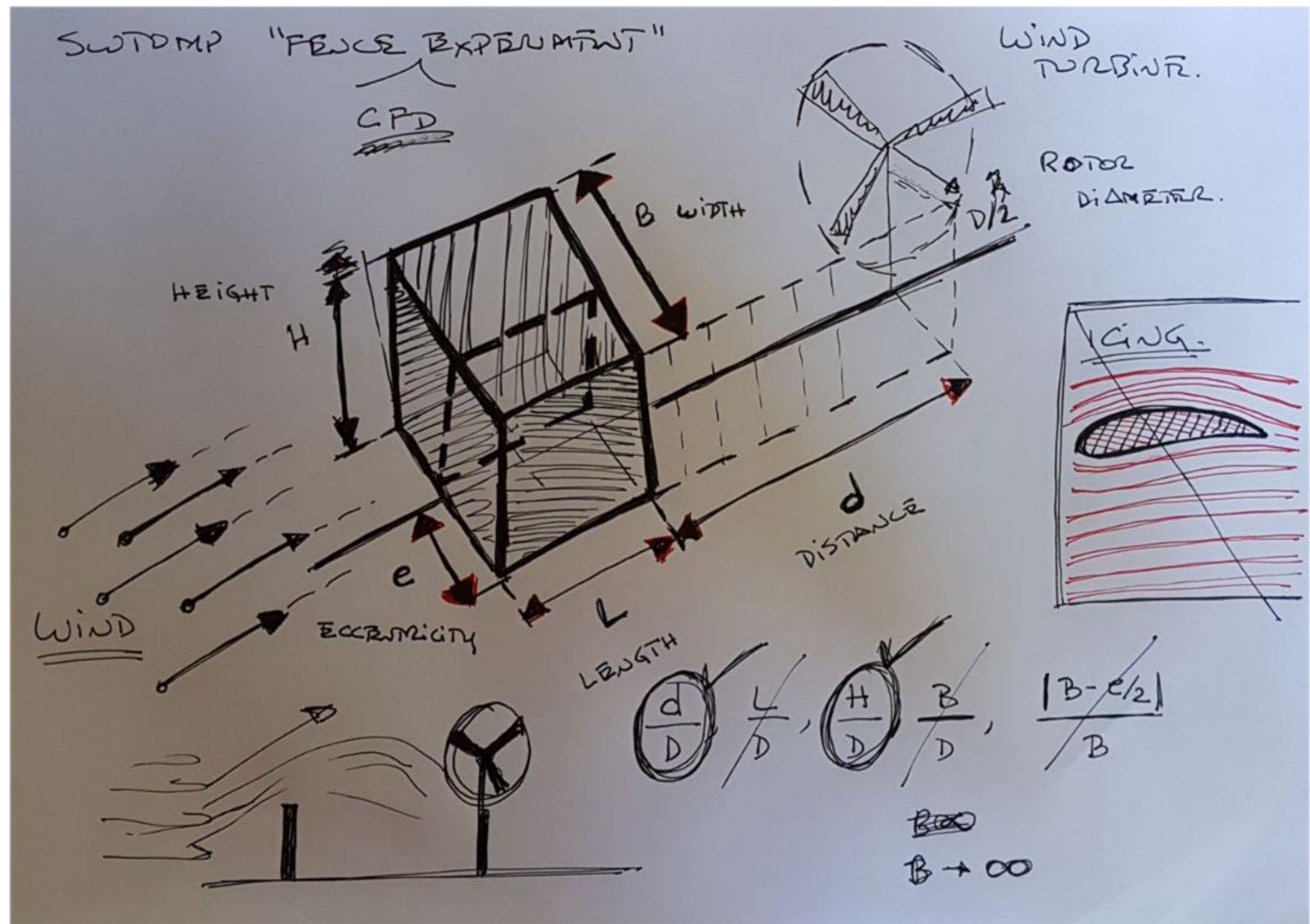
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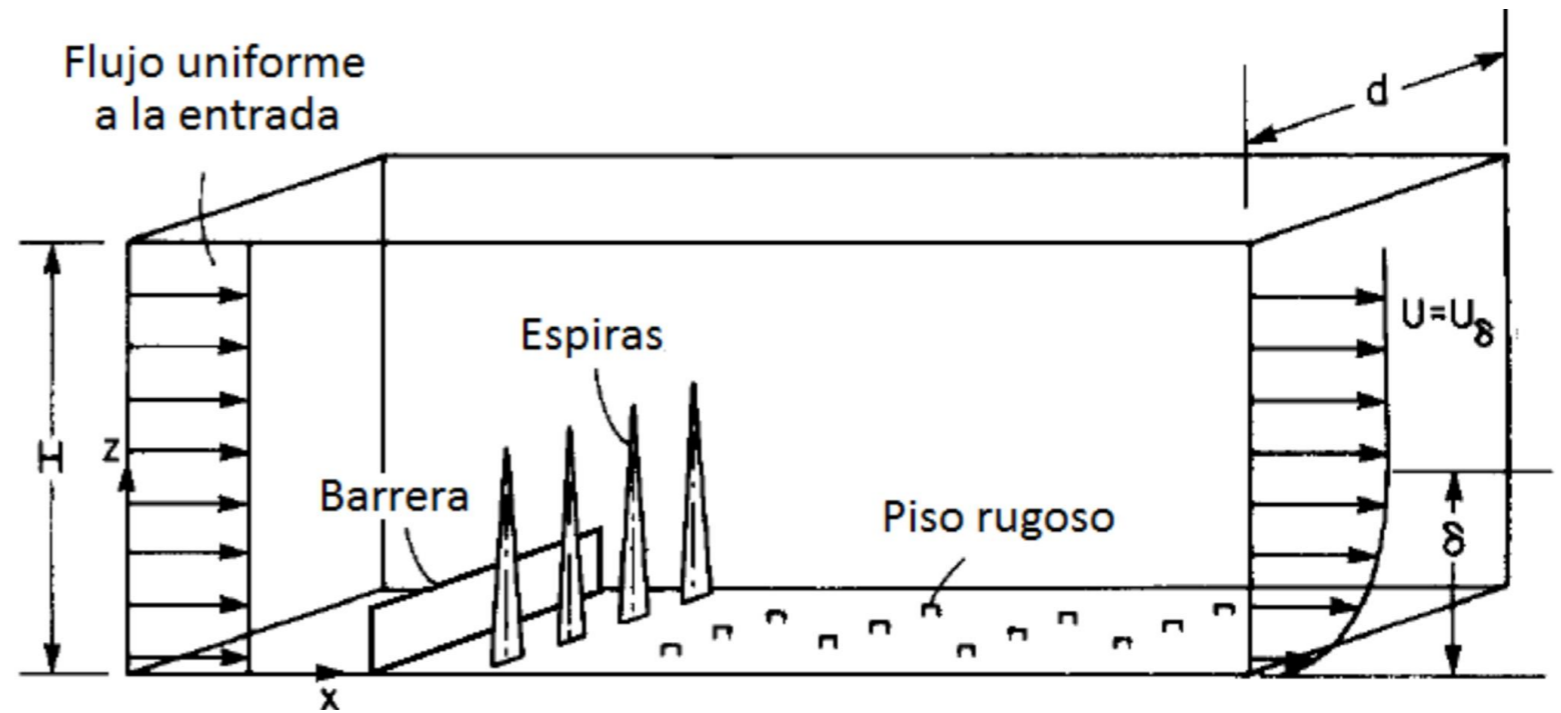
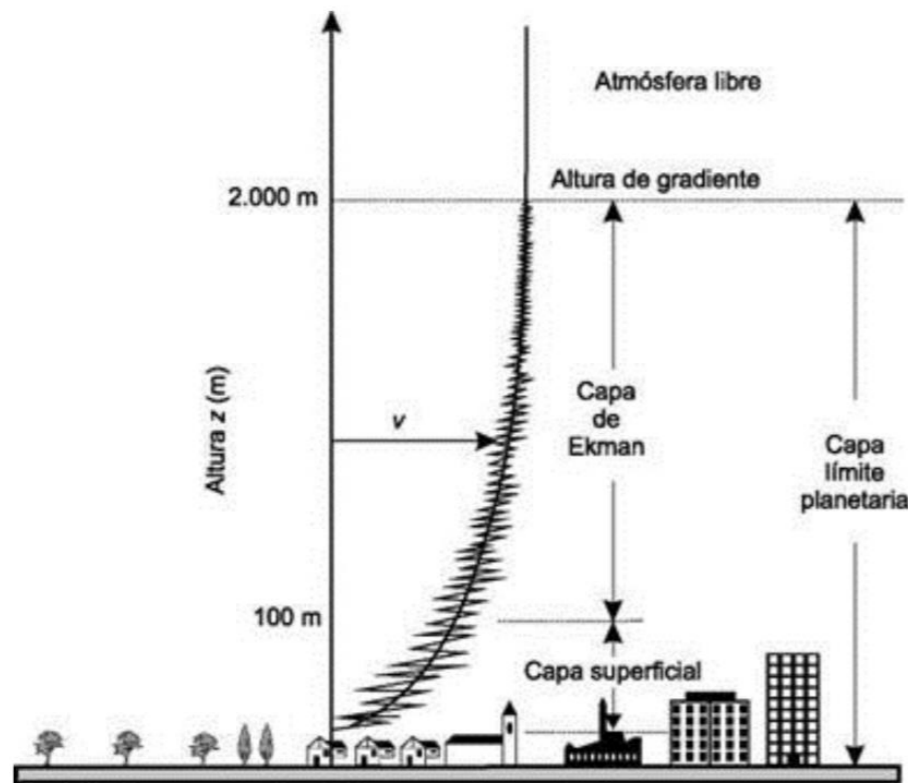
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When: Thursday, December 14, 2017 15:00-16:00 (UTC+03:00) Istanbul.

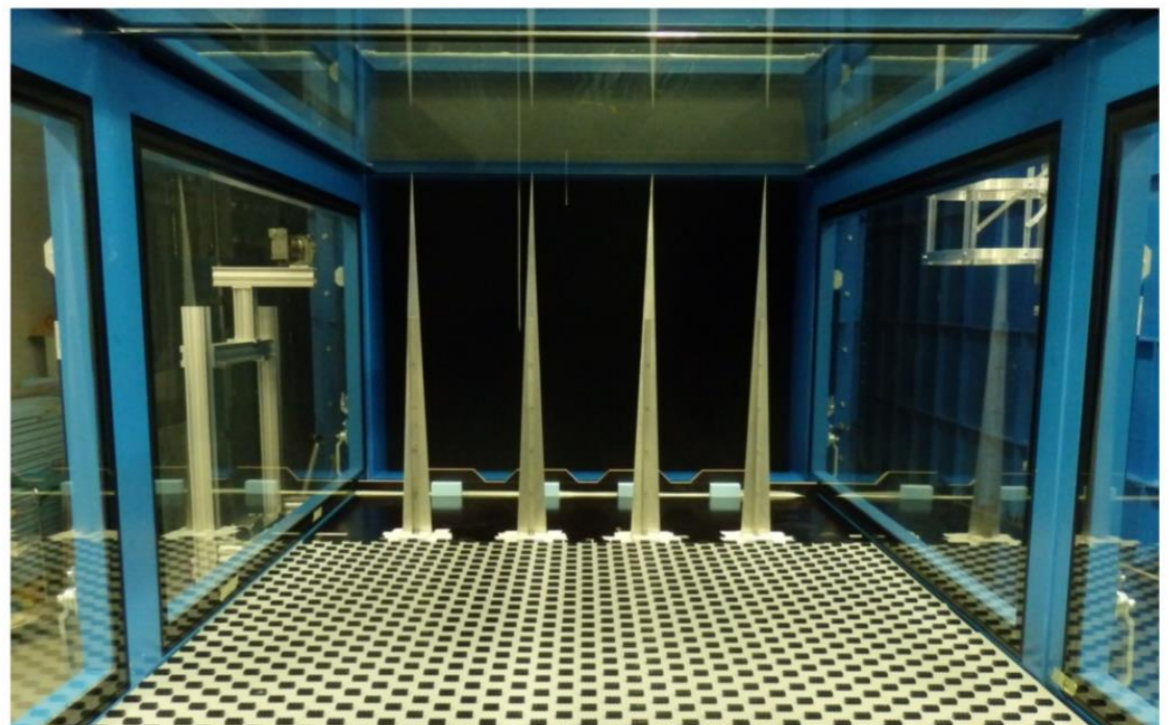
Where: <https://global.gotomeeting.com/join/885803509>



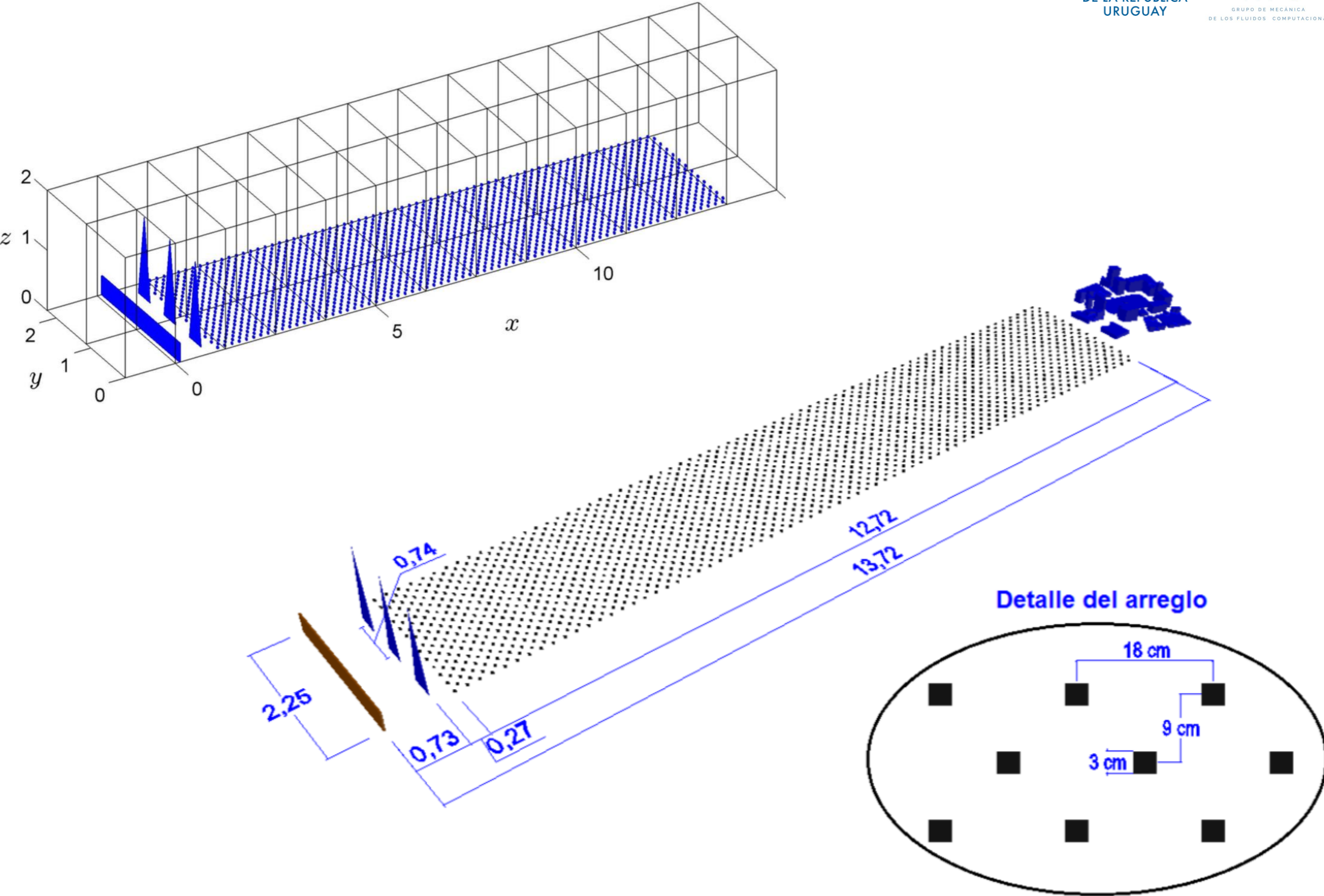
SWTOMP "Fence Experiment", ABL simulation



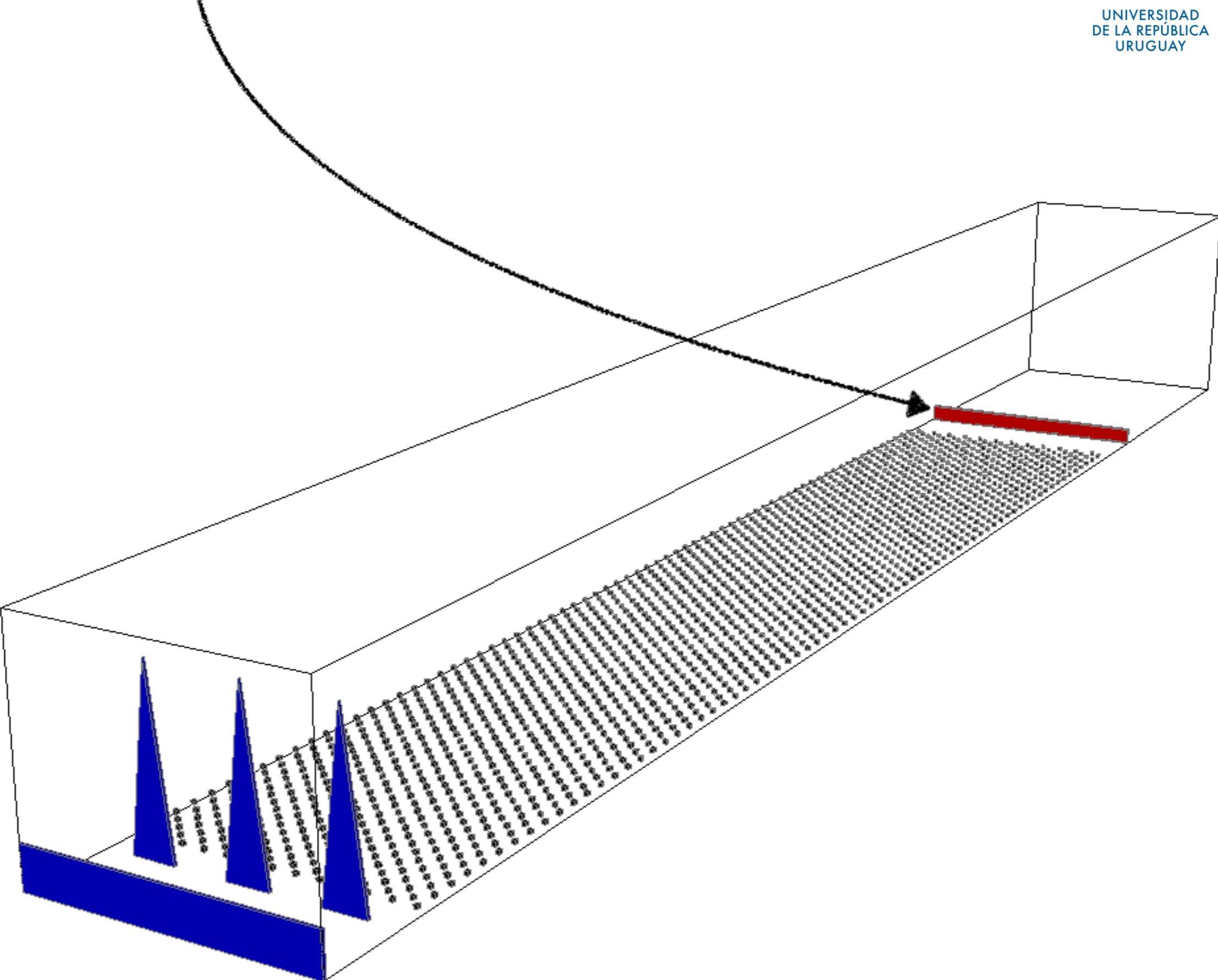
"The flow is tricked by the barrier into believing the fetch of roughness to be longer, and by the mixing-device that the barrier is not there at all!" [Cook (1978)]



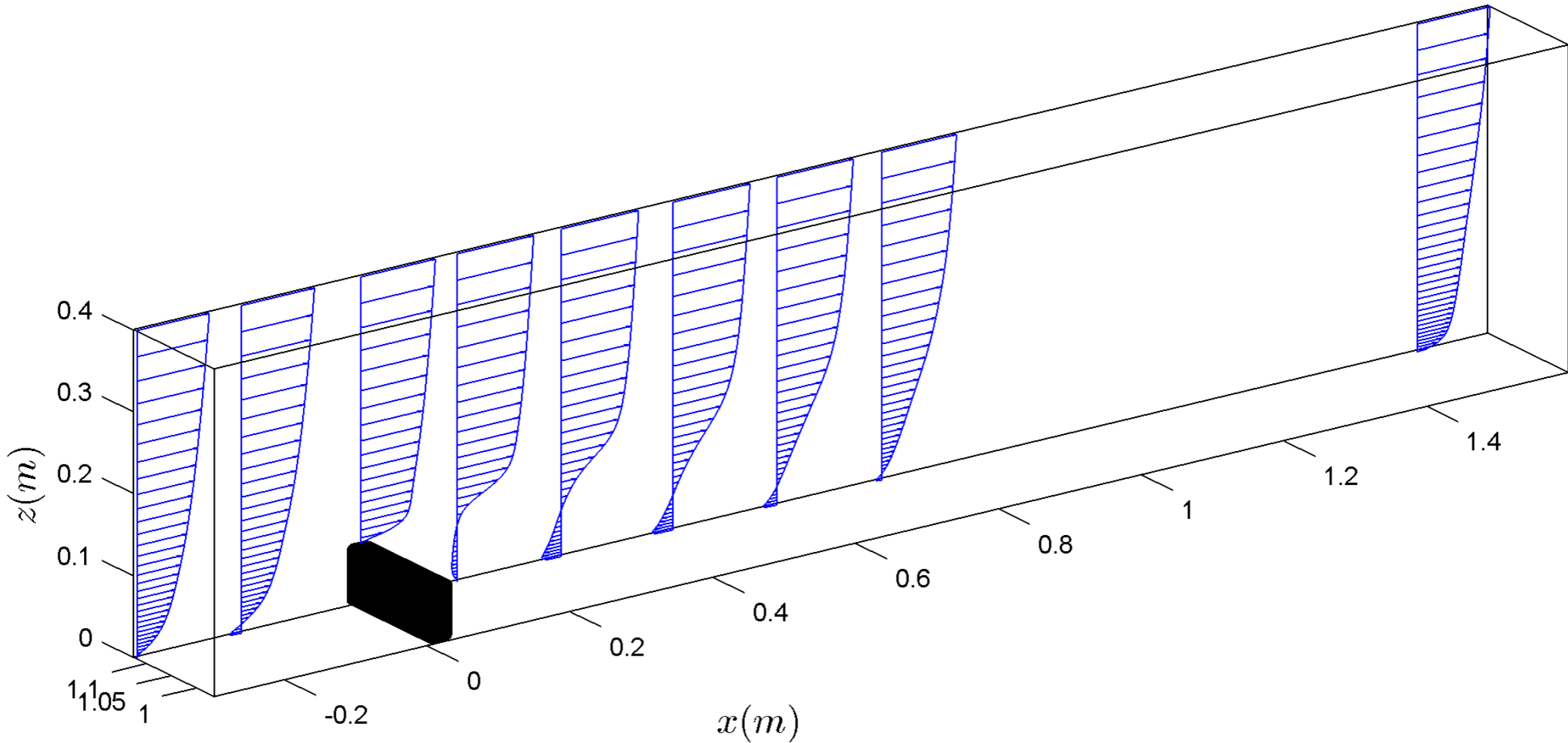
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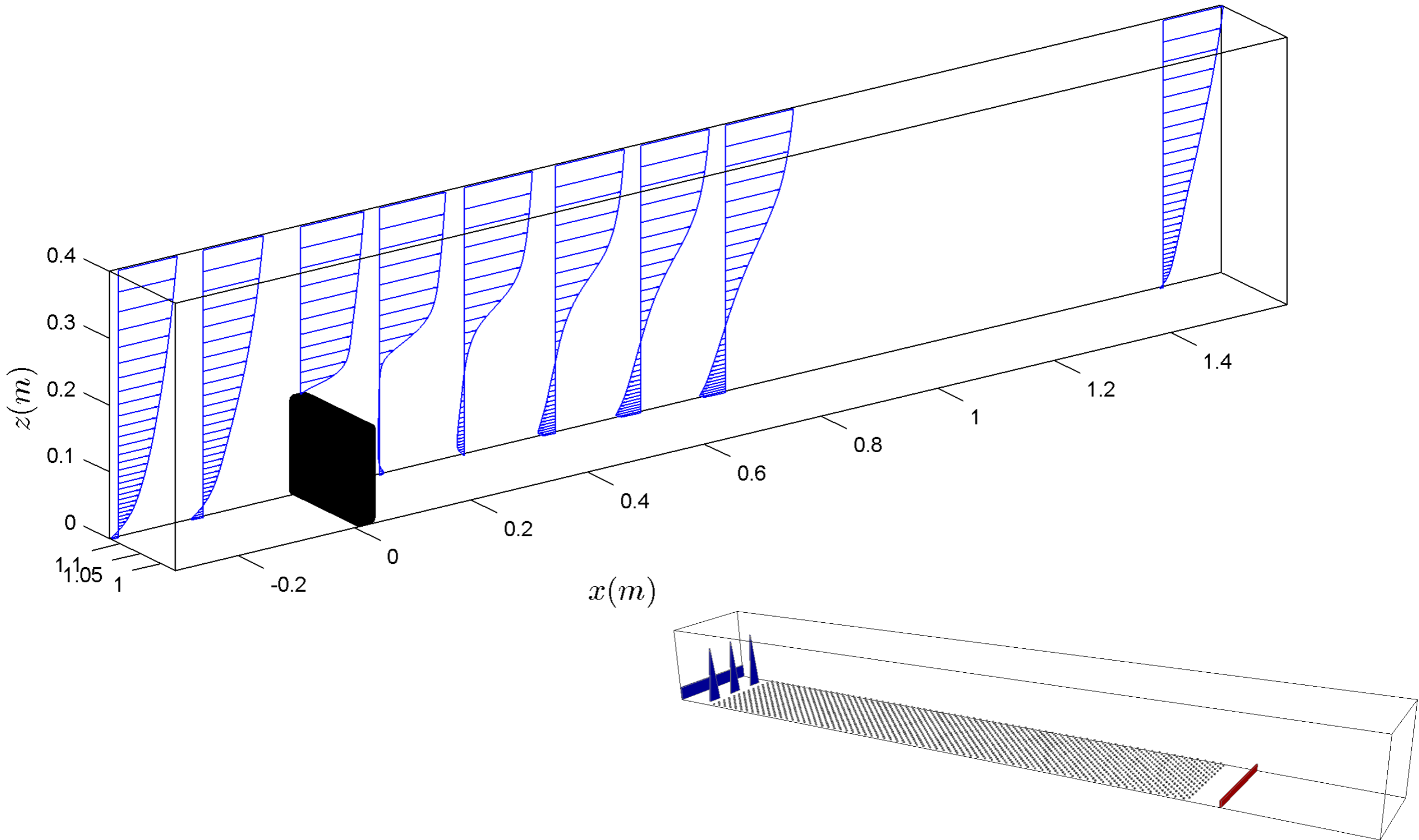
SWTOMP "Fence Experiment", ABL simulation (1:120)



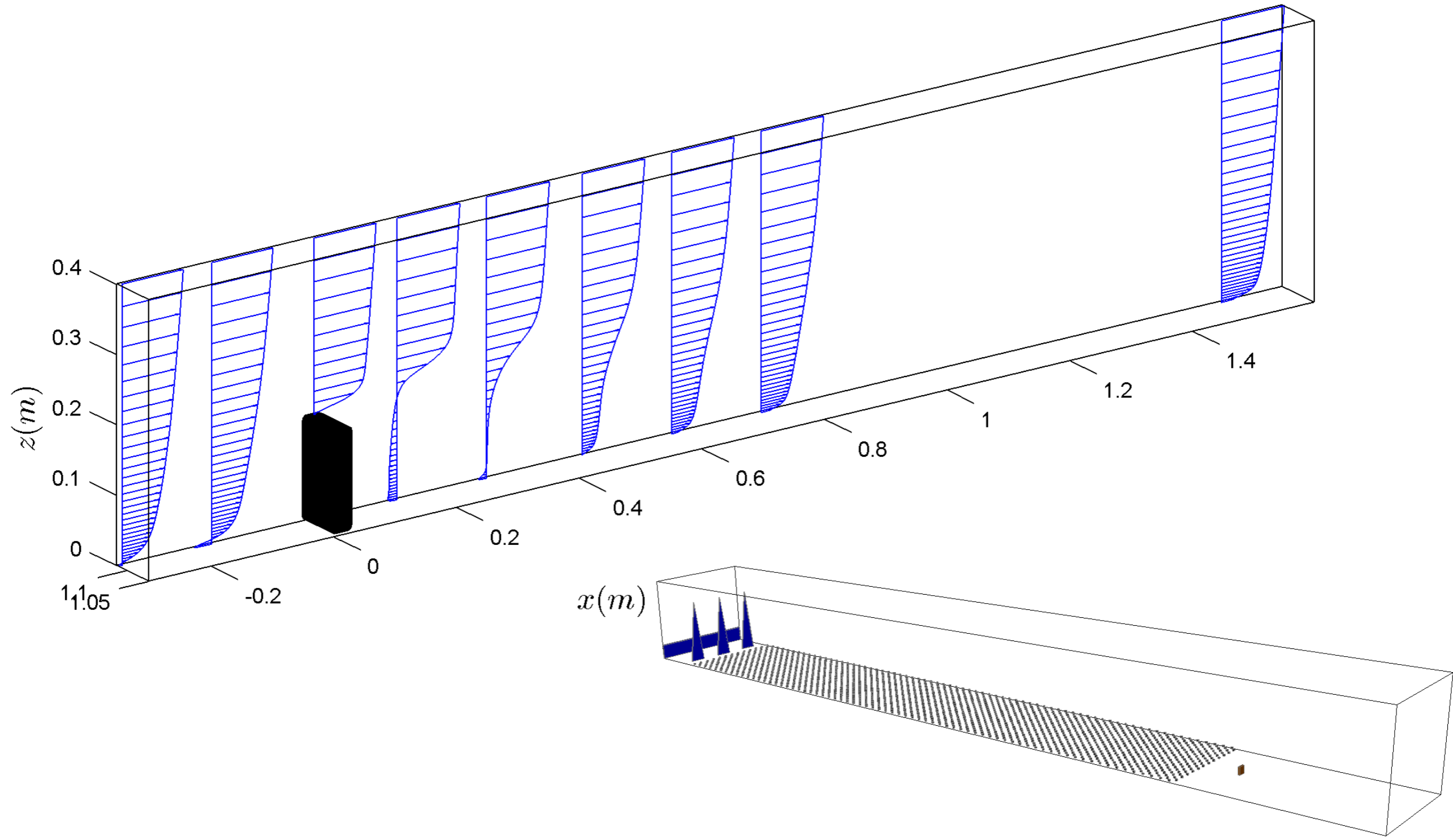
SWTOMP "Fence Experiment", ABL simulation (1:120)



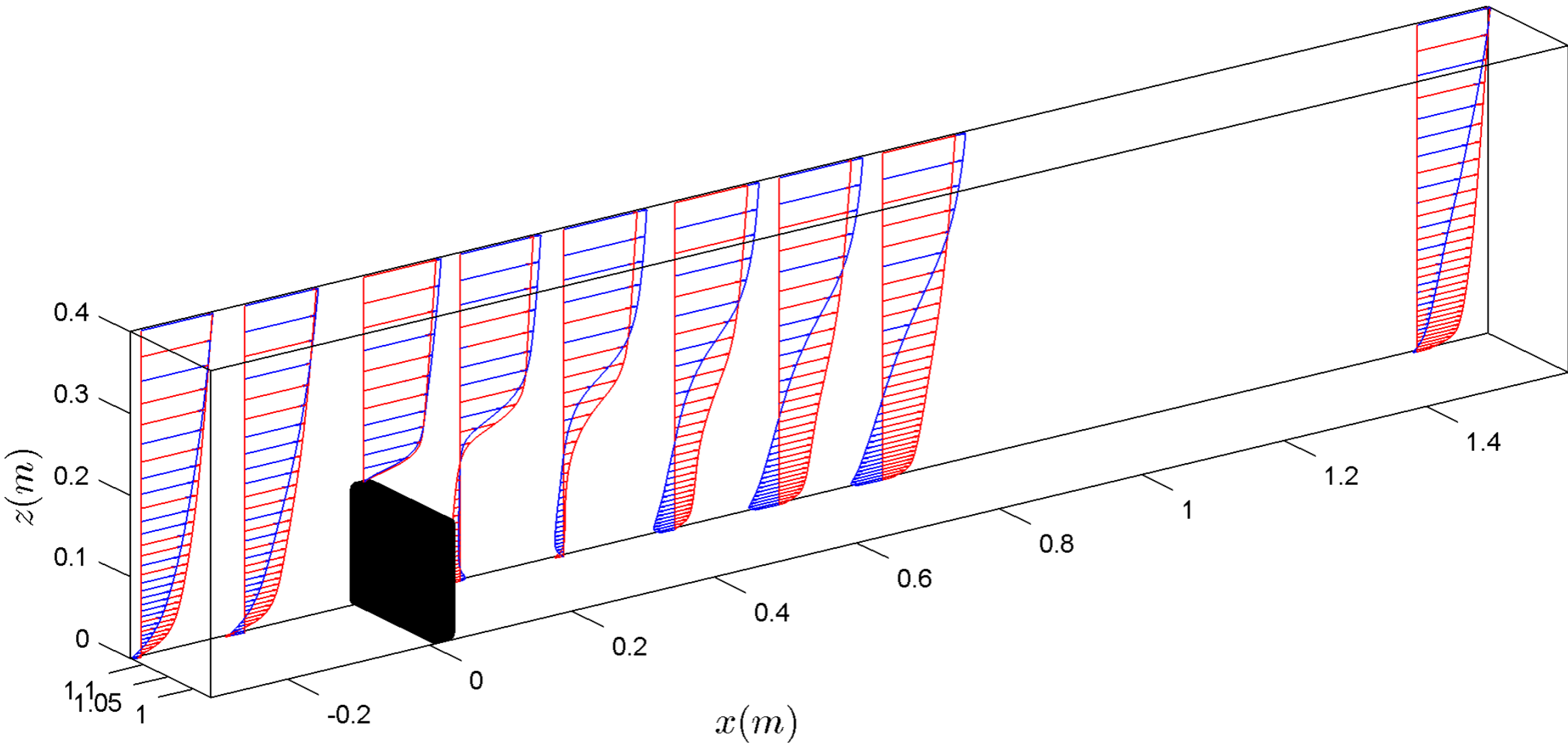
SWTOMP "Fence Experiment", ABL simulation (1:120)



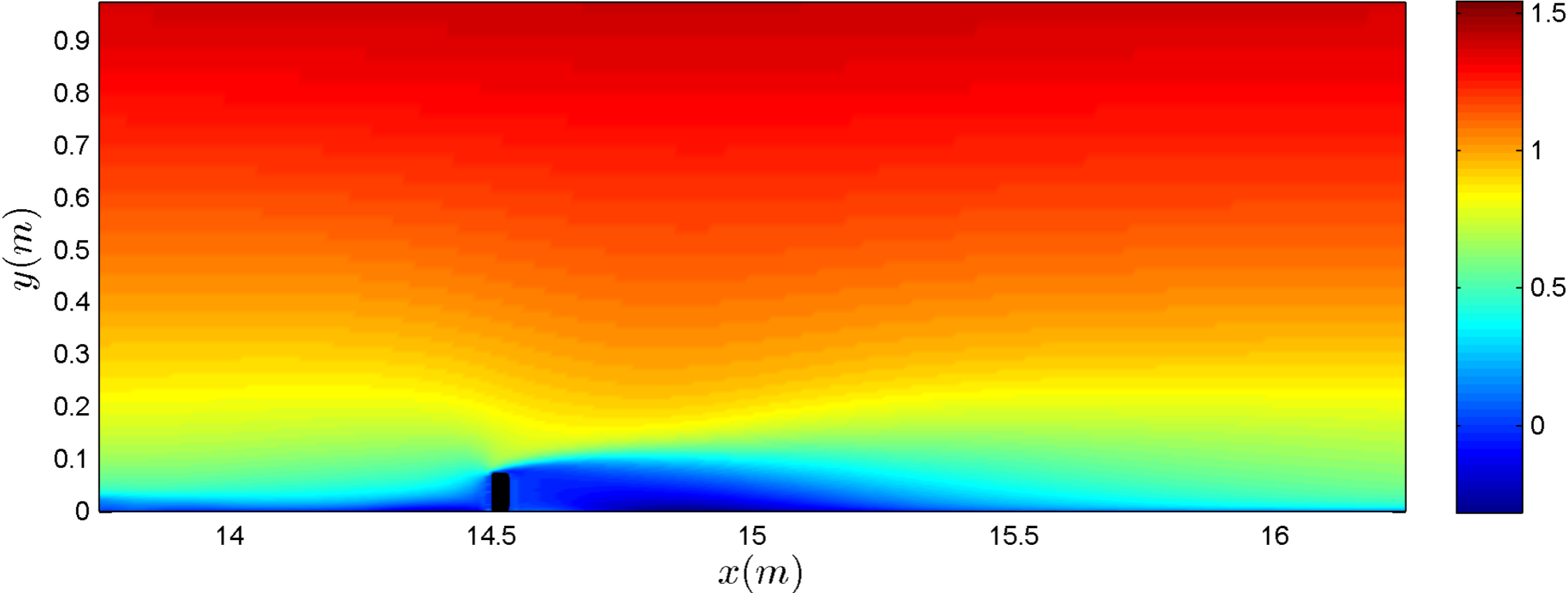
SWTOMP "Fence Experiment", ABL simulation (1:120)



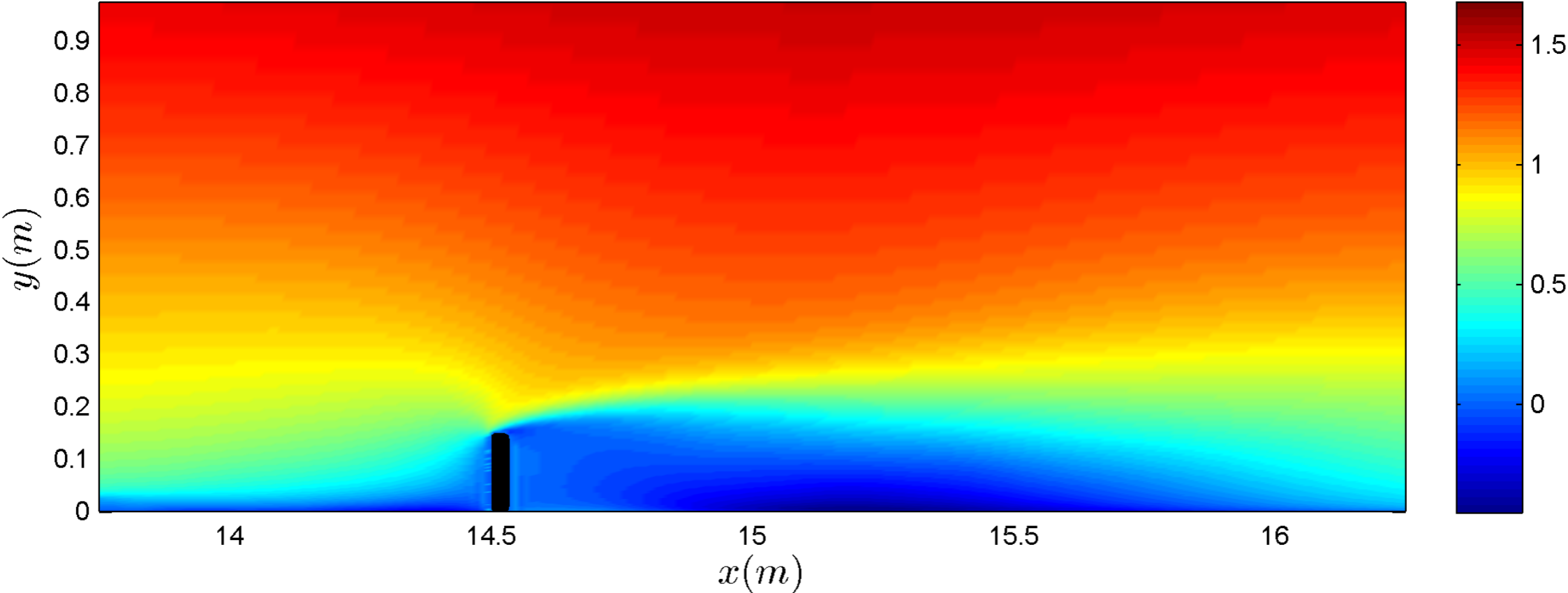
SWTOMP "Fence Experiment", ABL simulation (1:120)



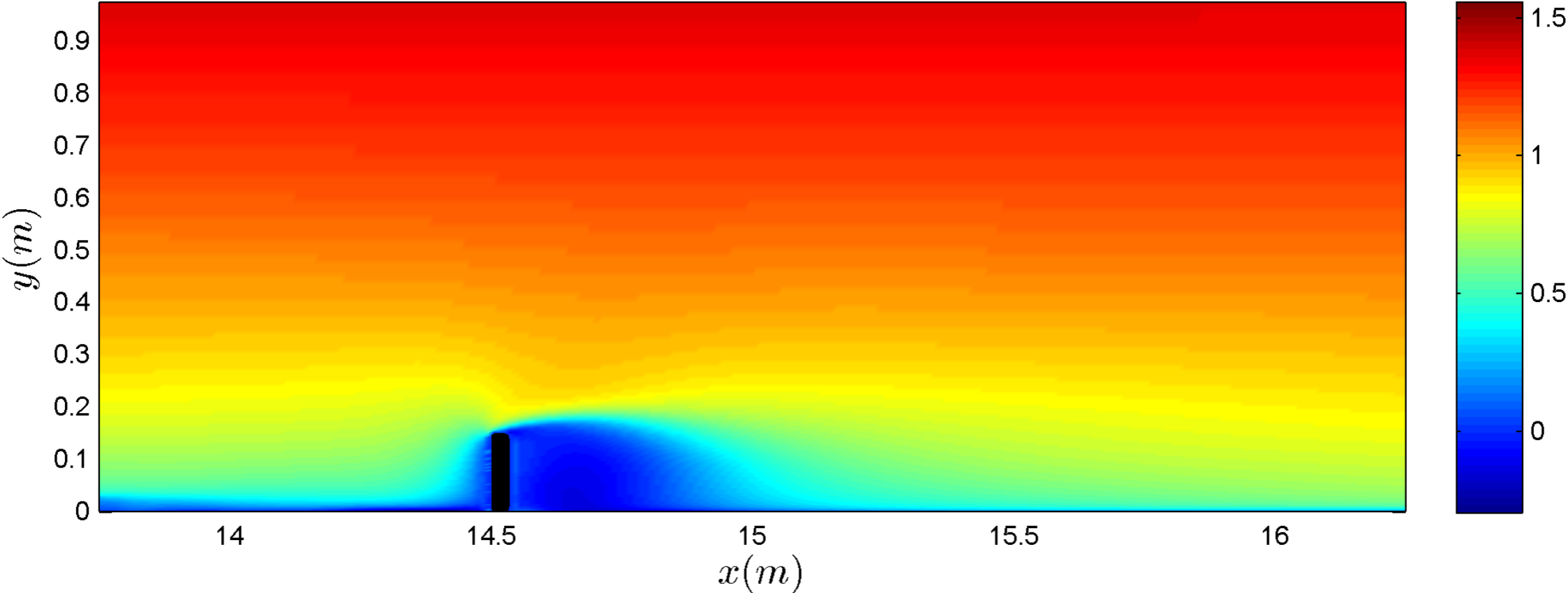
SWTOMP "Fence Experiment", ABL simulation (1:120)



SWTOMP “Fence Experiment”, ABL simulation (1:120)



SWTOMP “Fence Experiment”, ABL simulation (1:120)



Conclusions

**Powerful open source finite volume
based method for simulation of WTG**

**Capable of expansion through
modularity. Massive parallelism
through MPI**

**State of the art WTG methods on par
with international research and
industrial leading groups**

Next steps...

Complete migration to GPU - CUDA



Rotor resolved approach

Aeroelastic simulations



**Gracias!!
Gracias!!**

